

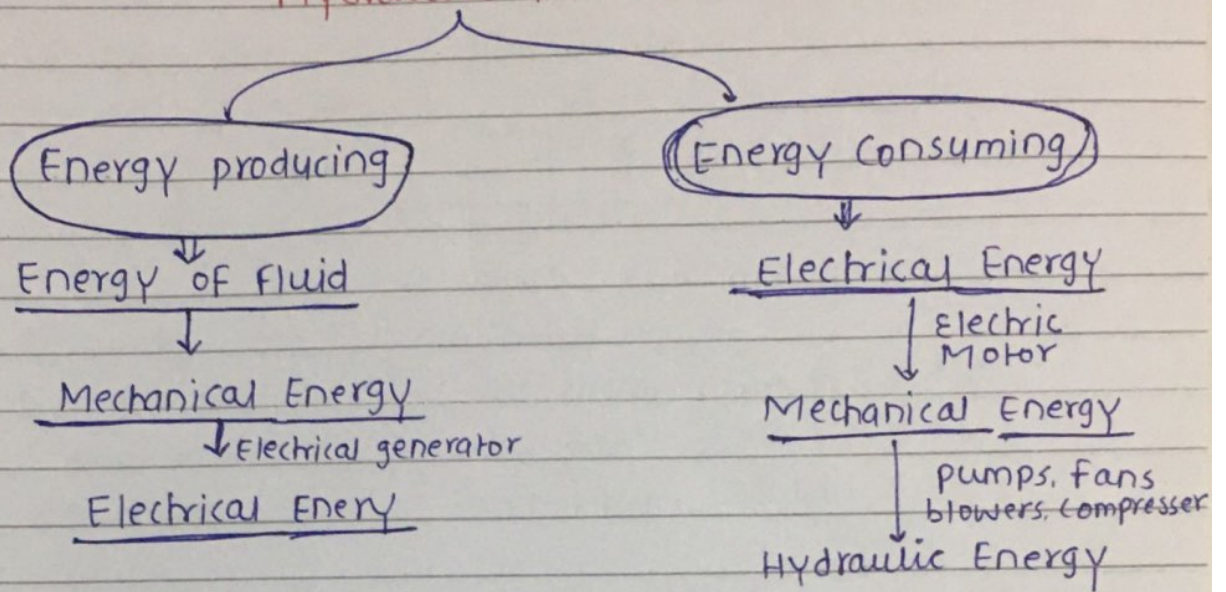
# HYDRAULIC MACHINES

6 Que.  
[PRE]

1. overview of hydroelectric projects (HEP's)
2. Hydraulic Turbines
  - Impulse turbines
  - Reaction Turbines.
3. Hydraulic pumps
  - centrifugal pump
  - Reciprocating pump.

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## Hydraulic Machines

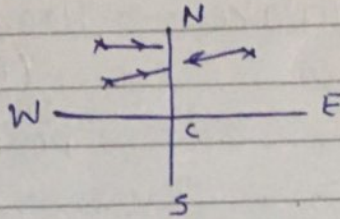


Ex: Hydraulic Turbine  
Steam Turbine  
Gas Turbine  
Wind Turbine



# ① OVERVIEW OF HEP'S

- 1) Generation
- 2) Transmission — PHCIL
- 3) Distribution.



Supply

500 MW  
500 MW  
500 MW

Demand

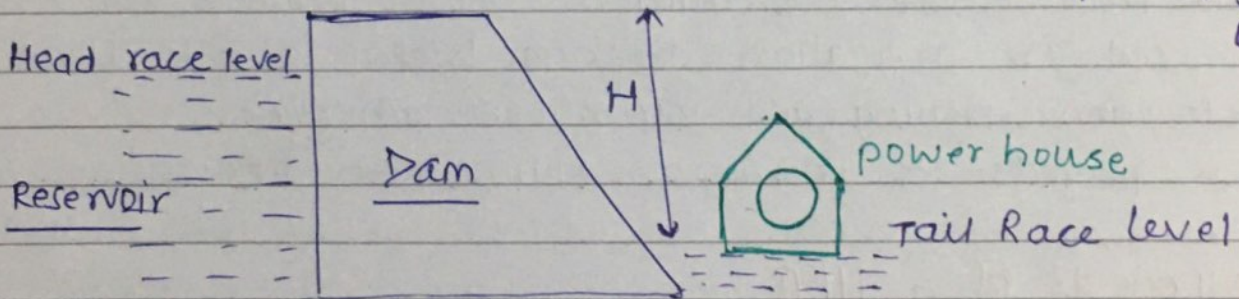
500 MW  $\Rightarrow f = 50 \text{ Hz}$   
600 MW  $\Rightarrow f < 50 \text{ Hz}$   
400 MW  $\Rightarrow f > 50 \text{ Hz}$

Demand  $\uparrow \rightarrow$  price  $\uparrow$   
Demand  $\downarrow \rightarrow$  price  $\downarrow$

$$N = \frac{60 \times f}{P}$$

$P \rightarrow$  pair of poles.

$N \rightarrow$  Speed of generator (rpm)



$$\text{Water power} = mgh = \rho Q g H$$

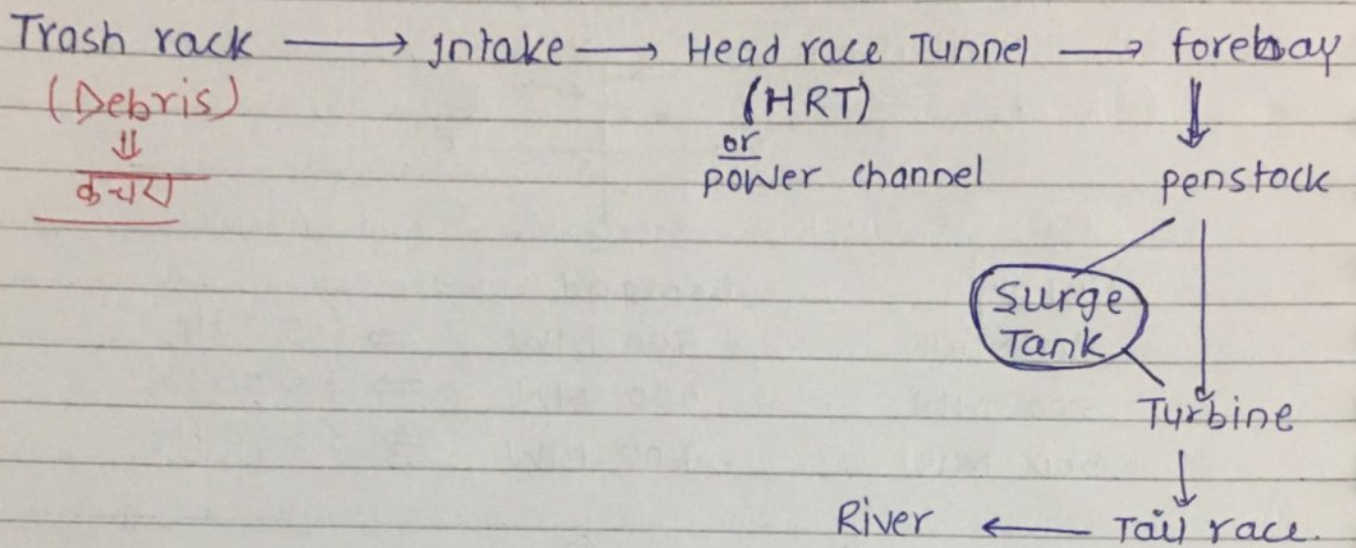
$$\text{load } \uparrow, P_{\text{gen}} \uparrow \Rightarrow Q \uparrow$$

$$\downarrow \rho g Q H$$

$\downarrow$  Varies, others are fixed.



## Waterways



In HEP, potential energy of water is utilised to drive a turbine which in turn runs the generator to produce electricity.

Apart from electricity production these projects can also be used for irrigation, drinking water supply, flood moderation, fishing and recreational activities.

Such projects are known as Multipurpose projects.

### Components of A HEP

① Reservoir Water available in catchment area is stored in reservoir so as to meet requirement of plant throughout the year.

Reservoirs can be natural as well as artificial.

Natural reservoirs are - lakes in mountains

Artificial reservoirs are made by constructing a dam across a river.

② Trash rack It is used to obstruct debris from going into the intake.



## NOTE

- ① Foreway and Surge tank, both serve same purpose
- ② Surge tank shall be located close to the turbine.

③ Waterway it is a passage through which water is carried from reservoir to the power house.

It consists of Head race Tunnel (HRT) or canal, Forebay, penstock, Surge tank.

(HRT)

④ Forebay It is an enlarged sec<sup>n</sup> @ the end of power channel. Its function is to store temporarily the water rejected by plant when the load is reduced and to meet the instantaneous increase demand of water due to sudden increase in load.

It also helps in absorbing the sudden rise in pressure due to closing of valves. When load on turbine ↓ Thus it prevents HRT from water hammer effect.

⑤ Pen stock It is a closed conduit for supplying water under pressure from forebay to turbine.

It is subjected to water hammer pressure due to fluctuation in turbine load. and to reduce this, a Surge TANK is provided.

⑥ Surge Tank It is a small reservoir fitted @ some opening in penstock to relieve the rejected flow when the load of turbine ↓ and thus it prevents penstock from water hammer effect.

⑦ Tail Race It is a waterway to carry water ~~from~~ discharged by turbine to a suitable point where it can be safely disposed off. or stored for pumping back into the original reservoir.

## Classification of HEP

Based on availability of Head.

① High Head plants:- Head  $> 250$  m.

② Medium Head plants:-  $30 < H < 250$  m.

③ Low head plants:-  $H < 30$  m.

This classification is not based on any scientific criteria.



### Based on load capacity

① Base load plant → Supply constant power

Ex' Thermal power plant, Hydropower plants with storage

② peak load plants Supply power during peak hours.

A storage type HEP is ideally suited for this purpose as it can be started in few minutes.

### Storage & pondage

storage & pondage of water is req<sup>d</sup> for regulation of flow of water so as to make it available in requisite ~~deficit~~ quantity to meet the power demand @ a given time.

Storage:- it is impounding of considerable amount of excess runoff during seasons of surplus flow for use in dry season. This is accomplished by constructing a dam across a river

pondage:- it is a regulating body of water in the form of relatively small pond or reservoir provided @ the plant. pondage is used to regulate the variable water flow to meet power demand.

Ex' forebay and Surge tank.

- Storage & pondage can be obtained from flow duration curve.

### Based on function

i) Run of river plant (ROR)

No storage of water is done and whatever water in the river is available is utilised to run the turbine.

ROR plants are suitable for perennial rivers.



- RoR plants can be with or without pondage.

With pondage  $\rightarrow$  peak load

Without pondage  $\rightarrow$  Base load

plants with pondage are provided with a Weir or barrage to accommodate sufficient storage to take care of load fluctuations.

ii) Storage plants These plants have storage provided by constructing Dam. These plants can meet peak load demands

iii) pumped storage plants

This type of plant is provided when there is a shortage of water.

These plants generate power during peak hours. During off peak hours water is pumped back from Tail race channel to the reservoir.

These plants use reversible turbines which functions as both turbine in peak hrs as well as pump during off peak hours.

Such plants convert low value of peak energy into high value on peak energy hence economically viable.

\* Suitable for peak load applications.

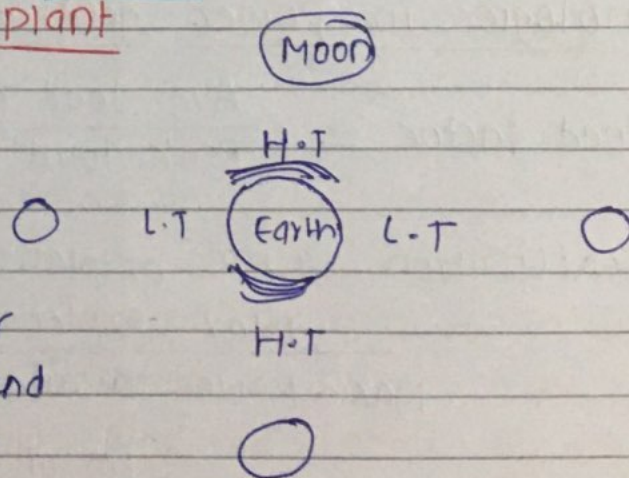
Based on Source of Energy

① river based power plant

② Tidal power plant

These plants tap the energy of earth's rotation.

There is rise in sea water during High tide period and Fall during L.T period.





Water rises and falls twice a day hence Head on turbine varies with time cyclically.

- During H.T when water level along coast rises, electricity can be produced.  
electricity can also be produced when ~~the~~ water is retreating by changing the dir<sup>n</sup> of rotation of turbine.

Turbines used are — Bulb turbine  
Tubular turbine.

### Advantages

- 1) Water
- 2) low Running cost
- 3) Renewable source of energy
- 4) NO GHG emissions
- 5) Quick starting & closure

↓  
Suitable for peak load applications.

### Limitations

- 1) Very high initial cost
- 2) long gestation period,
- 3) far from load centre

↓  
High transmission cost

- 4) resettlement & issues.
- 5) environmental flaws.

### Terminologies in power generation

(i) load factor = 
$$\frac{\text{Avg. load during a certain period}}{\text{Peak load during that period}}$$

(ii) Utilisation factor or plant use factor

$$\frac{\text{Max. power utilised}}{\text{Max. power available (= installed capacity)}}$$

(iii) Diversity factor = 
$$\frac{\text{Sum of individual max. demands}}{\text{Simultaneous Max. demand}}$$



$$\text{iv) Demand Factor} = \frac{\text{Simultaneous Max. Demand}}{\text{Sum of individual Max. demands}} \\ = \frac{1}{\text{Diversity Factor}}$$

(V) primary power (firm power)

It is the net amount of power that is available from a plant on a guaranteed basis.

vi) secondary power

excess power available above primary power during peak hours.

Qu. load on a hydal plant varies from a Min. of 10000 KW to a Max. of 35000 KW. Two Turbo-generators of capacity 22000 KW each have been installed. Calculate

i) Total installed capacity of plant

ii) Max. demand

iii) Load factor

iv) Utilisation or plant use Factor

i)  $2 \times 22000 = 44000 \text{ KW}$

ii)  $35000 \text{ KW}$

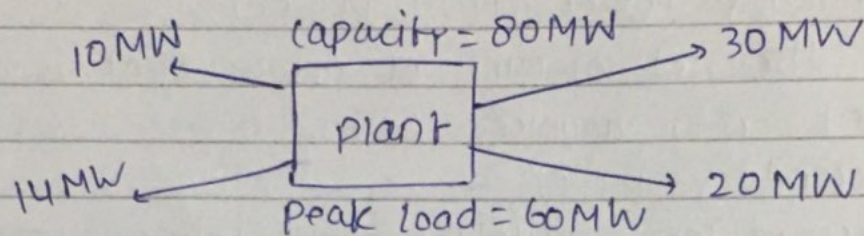
iii) 
$$\frac{10000 + 35000}{2} = 64.28\%$$

iv) 
$$\frac{35000}{44000} = 79.54\%$$

Qu. peak load on a powerplant is 60 MW. The loads having Max. demands of 30 MW, 20 MW, 10 MW and 14 MW are connected to power plant. capacity of power plant is 80 MW, Annual load factor is 0.5. Estimate.



- i) Avg. load on power plant
- ii) Energy supplied / year, in (kWh)
- iii) Demand factor
- iv) Diversity factor



i) load factor =  $\frac{\text{avg. load}}{\text{peak load}}$

$$0.5 = \frac{\text{avg. load}}{60}$$

$$\text{Avg. load} = 30 \text{ MW} \times 10^3$$

ii)  $30000 \text{ kW} \times 8760 = 262800 \text{ kW}\cdot\text{h}$

$$\{ 8760 = 365 \times 24$$

(iii)  $\frac{60}{74} = 0.81$

iv)  $\frac{74}{60} = 1.23$



## 2. HYDRAULIC TURBINES

Hydraulic Energy  $\xrightarrow{\text{Hydraulic Turbine}}$  Mechanical Energy  $\xrightarrow{\text{electrical generator}}$  Electrical Energy

Hydraulic Turbines are the devices which convert hydraulic energy into mechanical energy. This mechanical energy is further converted into electrical energy.

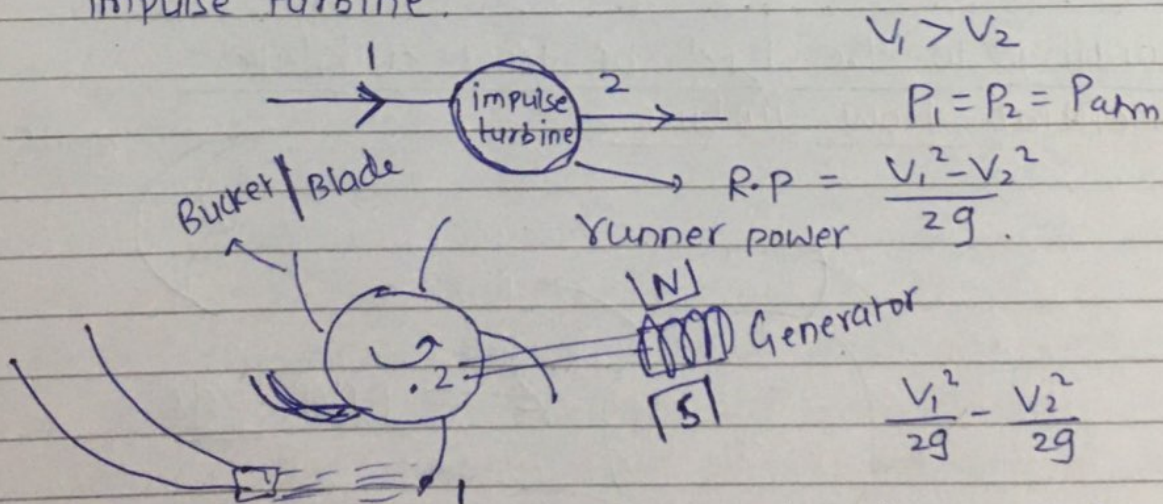
by a generator

### Classification of Hydraulic Turbines

① According to the action of water on moving blade.

A) Impulse Turbine, in impulse turbine, energy available @ entrance to the turbine is in the form of kinetic energy and water strikes the bucket with kinetic energy, causing an impact or ~~inter~~ impulse action. Throughout the movement of water through the runner, pressure remains atmospheric and kinetic energy of water is extracted in form of runner power.

Pelton wheel is the most commonly used impulse turbine.



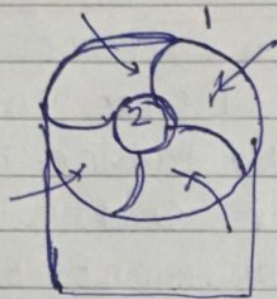
B) Reaction Turbine In reaction turbine runner blades are completely submerged in water and energy available @ the entrance to turbine is in the form of both p.E and K.E



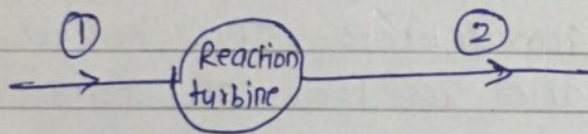
While flowing through runner, both KE & PE are extracted in form of runner power and @ exit both KE & PE are low.  
flow in reaction turbine is inside in air tight casing.

Commonly used reaction turbines are:—

Francis Turbine, Kaplan, Propeller Turbine



$$R.P = \left( \frac{P_1 - P_2}{\rho g} \right) + \left( \frac{V_1^2 - V_2^2}{2g} \right)$$

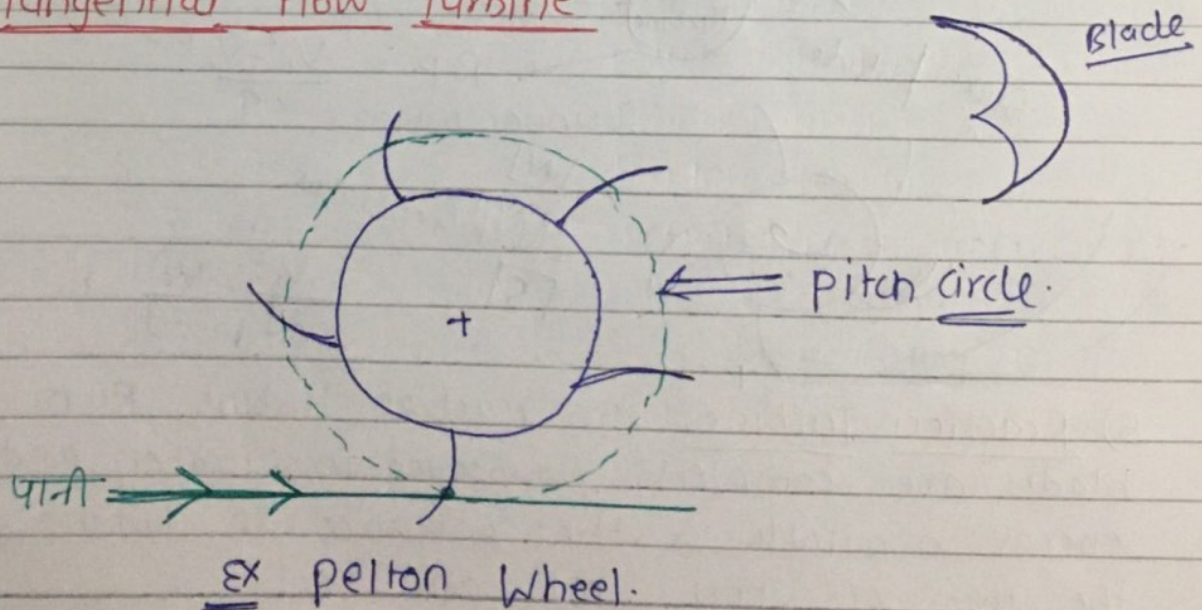


$$V_1 \gg V_2$$

$$P_1 \gg P_2$$

2) According to the Dir<sup>n</sup> of flow of water

① Tangential flow turbine





Water flows in a dir<sup>n</sup> tangential to the path of rotation of runner

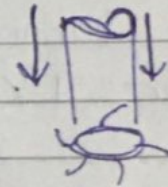
## ② Radial Flow

Water flows along the radial dir<sup>n</sup> and remains mainly in the plane normal to the axis of rotation of runner it can be i) inward Radial Flow:- Water enters @ outer circumference and flows radially inwards towards center of runner

Ex: old francis and Thomson turbine

ii) outward Radial Flow:- Water enters @ center and flows radially outwards towards the outer periferi of runner

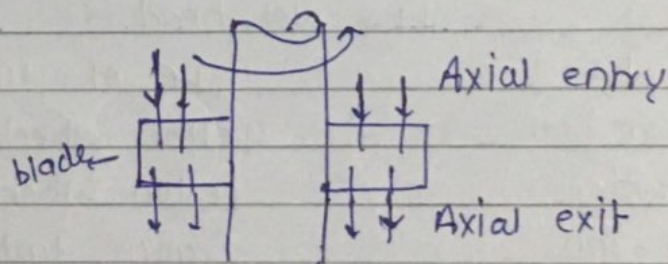
Ex: fourneyron Turbine



## ③ Axial Flow Turbine

In this, flow of water through runner is mainly along the dir<sup>n</sup> parallel to the axis of rotation of runner.

Ex: Kaplan Turbine, propeller turbine

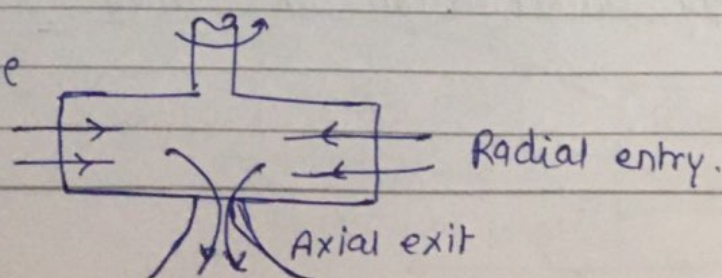


## ④ Mixed Flow Turbine

In this type of turbine Water enters the runner @ Outer periferi in radial dir<sup>n</sup> and leaves it at center in the dir<sup>n</sup> parallel to axis of rotation of runner

Ex: Modern

Francis turbine





### ③ Based on Head and discharge.

- ① pelton wheel → Works under high head hence requires low discharge

$$P = \rho g Q H$$

- ② francis turbine works under medium head and hence requires medium discharge.

- ③ Kaplan and propeller turbine work under low head hence require high discharge.

### ④ on Basis of specific speed ( $N_s$ )

specific speed is the speed of a geometrically similar turbine which while working under a head of 1m. produces 1 kW power

$$N_s = \frac{N \sqrt{P}}{(H)^{5/4}}$$

$N$  → speed of runner (rpm)

$P$  → power output in (kW)

$H$  → Net head

| $N_s$   | Type of turbine                  |
|---------|----------------------------------|
| 10-35   | pelton wheel with single jet     |
| >35     | pelton wheel with multijets      |
| 60-300  | francis turbine                  |
| 300-900 | Kapton turbine                   |
| >900    | Special type of runners are used |

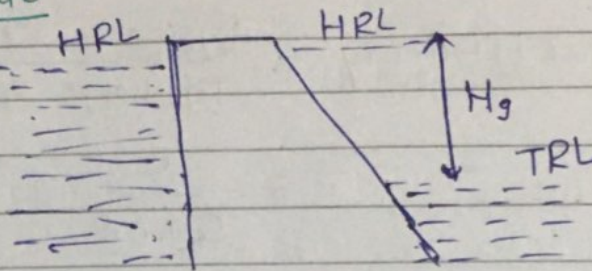
NOTE calculation of specific speed is independent of size of turbine hence this is the most scientific criteria for classification of turbine



## Head and efficiency of Turbine

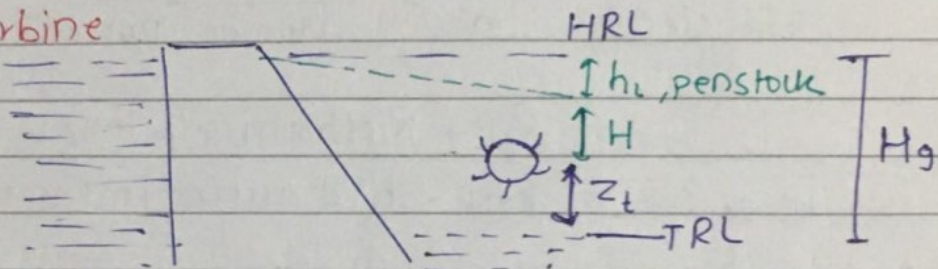
### ① Gross Head

It is the diff<sup>n</sup> of elevation b/w HRL and TRL



### ② NET Head for

Impulse Turbine



$$\text{Net head (H)} = H_g - h_{L, \text{penstock}} - Z_t$$

$Z_t \rightarrow$  height of turbine above TRL

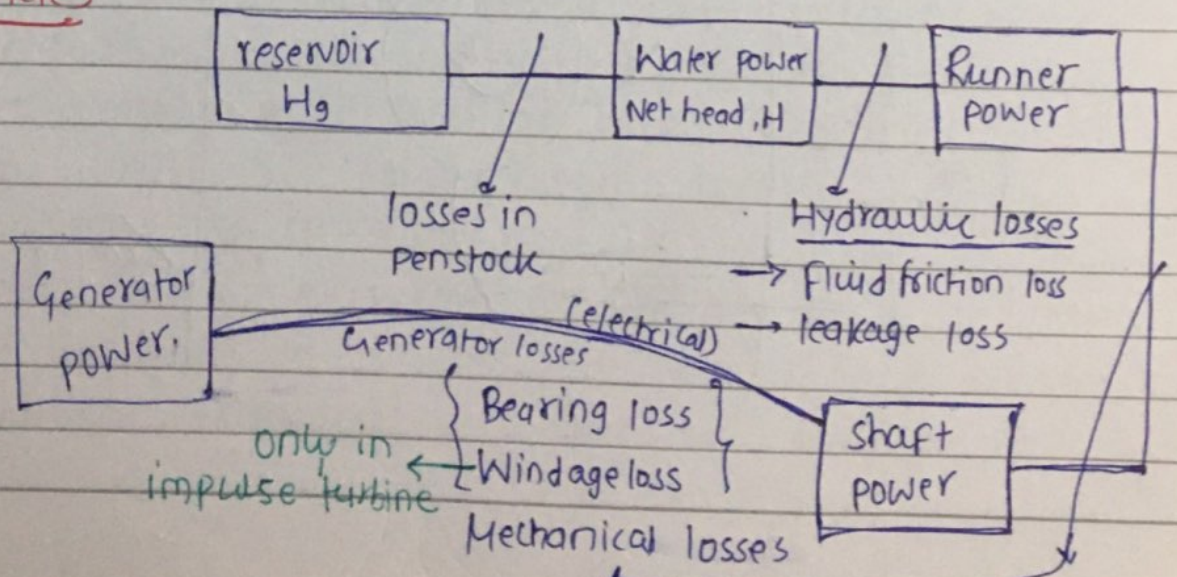
for

Reaction turbine

$$H = H_g - h_{L, \text{penstock}} - \frac{V_{dt}^2}{2g}$$

$V_{dt} \rightarrow$  velocity @ exit of draft tube

## Efficiencies





$$\text{Hydraulic Efficiency } \eta_h = \frac{\text{Runner power}}{\text{Water power}} = \frac{RP}{\rho g Q H}$$

$$\begin{aligned} \text{Volumetric Efficiency} &= \frac{\text{Discharge striking the Runner}}{\text{Discharge Supplied to the Runner}} \\ &= \frac{Q}{Q + \Delta Q} \quad \Delta Q \rightarrow \text{leakage loss} \end{aligned}$$

$$\text{Mechanical Efficiency } \eta_m = \frac{\text{Shaft power}}{\text{Runner power}}$$

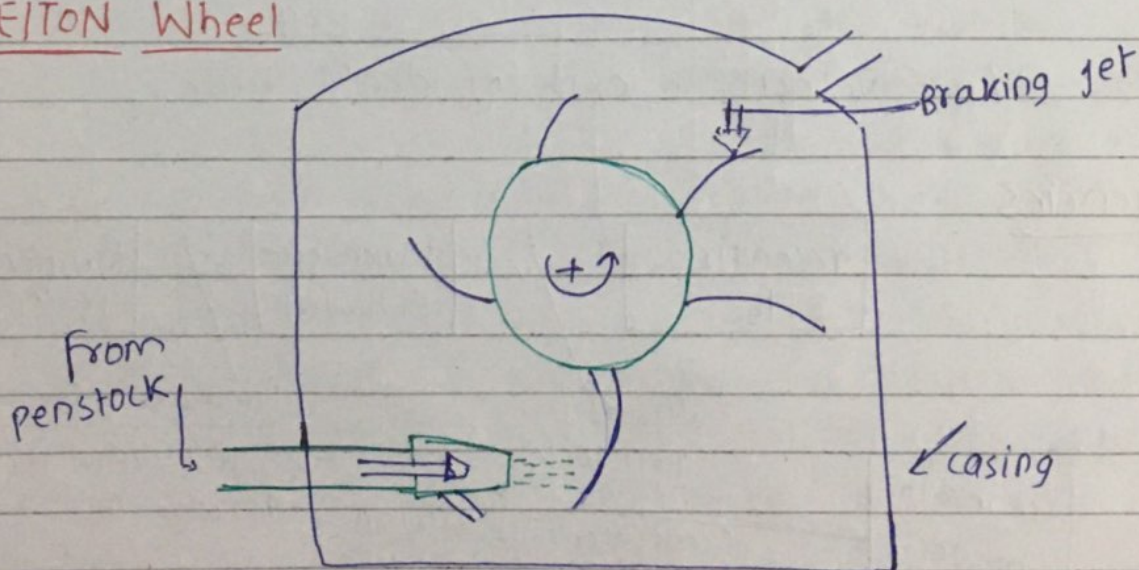
$$S.P = R.P - \text{mechanical losses}$$

$$R.P = W.P - \text{hydraulic losses}$$

$$\text{Overall Efficiency } \eta_o = \frac{\text{Shaft power} \rightarrow \text{Brake horse power (BHP)}}{\text{Water power} \rightarrow \text{Water horse power (WHP)}}$$

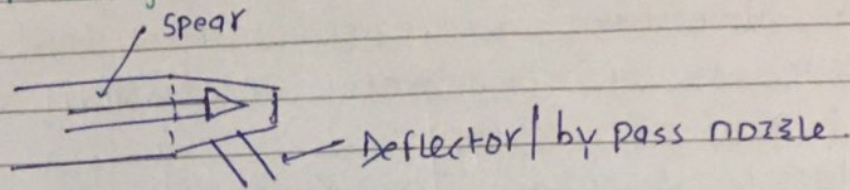
$$\eta_o = \eta_h \times \eta_m$$

### PELTON Wheel





### Nozzle spear arrangement



load  $\uparrow \Rightarrow P_{gen} \uparrow \Rightarrow Q \uparrow$  } Governing.  
load  $\downarrow \Rightarrow P_{gen} \downarrow \Rightarrow Q \downarrow$  }

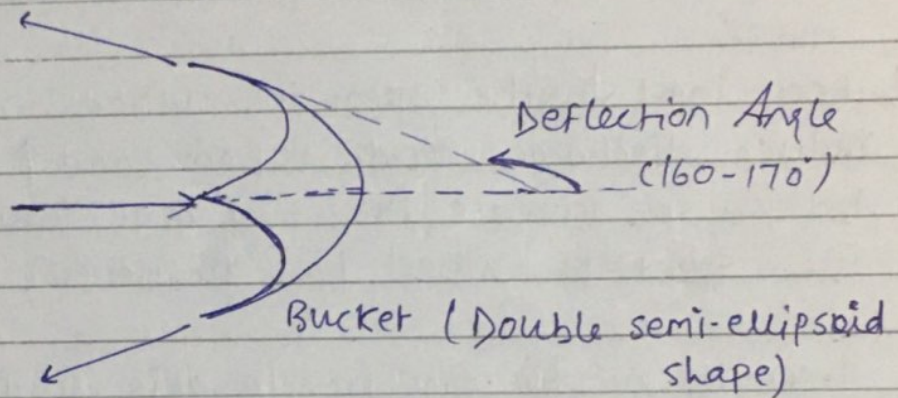
→ High  $H$ , low  $Q$

→ Tangential flow

→ Impulse

→  $V_2 \ll V_1$

→  $P_1 = P_2 = P_{atm}$



### Major components of a pelton wheel

- 1) NOZZLE spear arrangement
- 2) Bucket
- 3) Casing
- 4) Braking jet

pelton wheel turbine is a tangential flow, high head impulse turbine. in this turbine water jet impinges on bucket attached to the shaft of turbine which causes force on bucket. This force creates torque necessary for rotation of shaft.

all the head is converted to kinetic head before the jet impinges on bucket. and the kinetic energy of water is extracted in form of runner power



While moving on bucket, water is always in contact with atmosphere and hence throughout its operation water is at atmosphere pressure.

### Major components.

#### ① Nozzle & spear Arrangement

It is a guide mechanism which controls quantity of water passing through nozzle and striking the bucket. It helps in meeting the variable demand of power.

When load  $\downarrow$  the spear is pushed into nozzle to reduce discharge and when load  $\uparrow$  spear is pulled back to increase the discharge. Thus it maintains the speed of wheel (or frequency) constant.

Deflector or by pass nozzle is used to control the water hammer pressure rise due to sudden reduction in discharge.

#### ② Bucket

It has the shape of double semi-ellipsoid. Advantage of having this shape is that net axial thrust gets reduced.

#### ③ Casing

In Pelton wheel casing doesn't perform any hydraulic function. Its purpose is to prevent the splashing of water and to guide the water to the tail race. It also safeguards against accidents.

#### ④ Braking jet

It is provided to stop the runner in shortest possible time. It is in the form of a jet striking the bucket from opposite dir<sup>n</sup>.

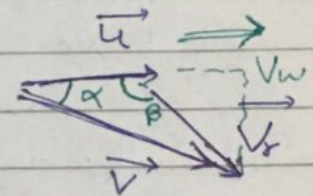
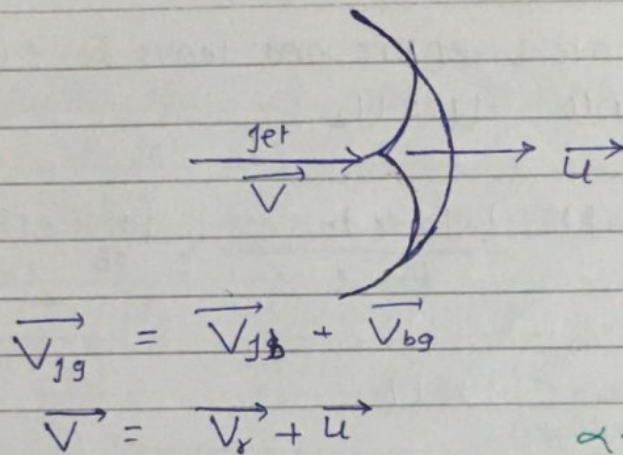
NOTE  $\Downarrow$  Not used for speed reduction, only used to stop the runner.



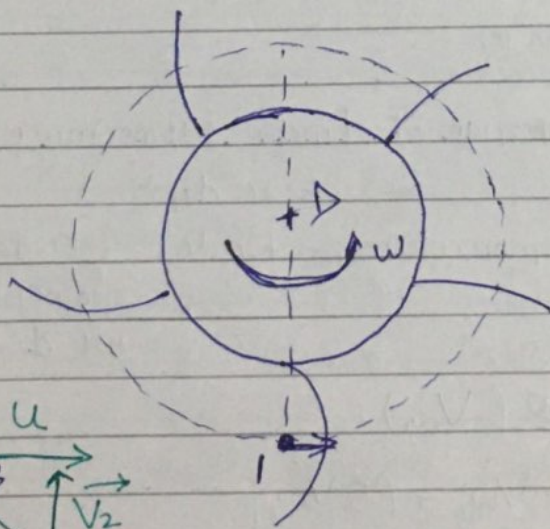
NOTE • Ordinarily pelton wheel has single jet and horizontal shaft. However no. of jets depends on specific speed. for specific speed  $> 35$ , Generally multijets are used.

- World wide Max. 6 jets have been employed without interference.

### power output in pelton wheel



$\alpha \rightarrow$  Angle b/w  $\vec{V}'$  &  $\vec{u}$   
 $\beta \rightarrow$  Angle b/w  $\vec{V}_r$  &  $-\vec{u}$

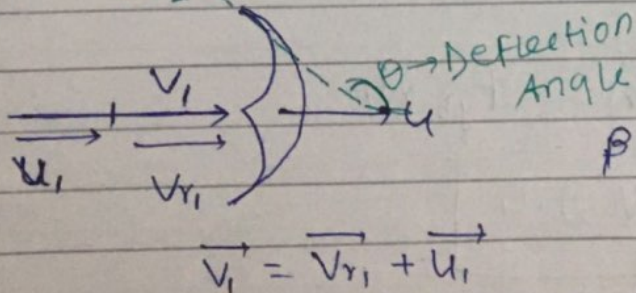


$D \rightarrow$  Pitch circle diameter  
 $u \rightarrow$  peripheral tangential velocity of wheel.

$$= \omega \times D/2$$

$$= \frac{2\pi N}{60} \times D/2$$

$$u_1 = \frac{\pi D N}{60} = u_2$$



$\beta \rightarrow$  Angle b/w  $\vec{V}_r$  &  $-\vec{u}$   
 $= 180^\circ$



- $V_1$  and  $V_2$  are absolute velocity of jet at inlet and ~~exit~~ exit
- $V_{r1}$  &  $V_{r2}$  are the relative velocities of jet w.r.t bucket at inlet & exit
- $u$  → tangential velocity of bucket
- $V_{w1}$  and  $V_{w2}$  → tangential component of absolute velocity at inlet and exit.

In pelton wheel water enters and leaves @ same Diameter ( $D$ ) hence  $u_1 = u_2$

$$\text{Power output} = \frac{\text{Work done}}{\text{time}} = \frac{\text{force} \times \text{Displacement in dir of force}}{\text{time}}$$

$$= \frac{\text{force}}{\text{Jet on bucket}} \times u$$

$$F_{\text{Jet on bucket}} = -F_{\text{bucket on jet}}$$

$$F_{\text{bucket on jet}} = \text{Rate of change of Linear Momentum of jet in } +x \text{ dir}^n$$

$$= \text{Final Linear momentum in } +x \text{ dir}^n - \text{Initial Linear momentum in } +x \text{ dir}^n$$

$$F_{bj} = \rho Q (-V_{w2}) - \rho Q (V_{w1})$$

$$F_{\text{Jet on bucket}} = -F_{bj} = -[-\rho Q V_{w2} - \rho Q V_{w1}]$$

$$= \rho Q (V_{w1} + V_{w2})$$

$$\text{Power output (Runner power)} = F_{jb} \times u$$

$$\boxed{R.P = \rho Q (V_{w1} + V_{w2}) \cdot u}$$



$Q \rightarrow$  Discharge hitting the bucket

$$Q = A \times V_1$$

$A \rightarrow$  Area of jet

$$= \frac{\pi}{4} d^2$$

where  $d \rightarrow$  dia of jet

$V_1 \rightarrow$  absolute velocity of jet

\* if it was a single jet, we would have taken  $V_{r1}$

It is assumed that at every instant, there is a bucket in front of jet.

Fulcr's head ( $H_e$ )

$$H_e = \frac{\text{Runner power}}{\text{Wt. of fluid}} = \frac{\rho Q (V_{w1} + V_{w2}) u}{\rho Q g}$$

$$H_e = \frac{(V_{w1} + V_{w2}) u}{g}$$

$$R.P = \rho Q (V_{w1} + V_{w2}) u$$

From velocity  $\Delta^u$

$$V_{w1} = V_1$$

$$V_{w2} = V_{r2} \cos \beta - u$$

$$K \rightarrow \text{bucket friction coefficient} = \frac{V_{r2}}{V_{r1}}$$

$$V_{r2} = K V_{r1}$$

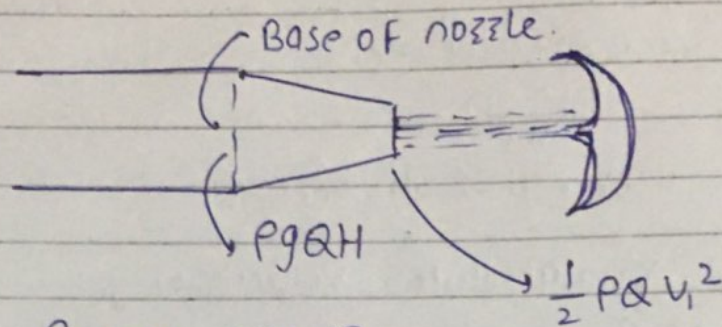
$$V_{r2} = K (V_1 - u)$$

$$V_{w2} = K (V_1 - u) \cos \beta - u$$

$$R.P = \rho Q [(V_1 - u) (1 + K \cos \beta)] u$$



## Hydraulic Efficiency ( $\eta_h$ )



$$\eta_h = \frac{\text{Runner power}}{\cancel{\text{No}} \frac{1}{2} P Q V_1^2}$$

$$\eta_h = \frac{\text{Runner power}}{P g Q H} \Rightarrow \text{When nozzle is considered as an integral part of turbine.}$$

$$\eta_h = \frac{P Q (V_1 - u) u (1 + K \cos \beta)}{P g Q H}$$

$$\eta_h = \frac{(V_1 - u) u (1 + K \cos \beta)}{g H}$$

for max.  $\eta_h$

$$\frac{d\eta_h}{du} = 0 \Rightarrow V_1 - 2u = 0$$

$$u = \frac{V_1}{2} ***$$

$$(\eta_h)_{\max} = \frac{(V_1 - \frac{V_1}{2}) \frac{V_1}{2} (1 + K \cos \beta)}{g H}$$

$$\eta_{h, \max} = \frac{V_1^2 (1 + K \cos \beta)}{4 g H}$$

$V_1 \rightarrow$  absolute velocity of jet

$$V_1 = K v_1 \sqrt{2 g H}$$



$$\beta = 180 - \theta$$

$K_{v_1} \rightarrow$  coefficient of Velocity of nozzle.

$$K_{v_1} \Rightarrow 0.97 - 0.99$$

$$\eta_{h, \max} = \frac{K_{v_1}^2 \cdot 2gH (1 + K \cos \beta)}{4gH}$$

$$= \frac{K_{v_1}^2 (1 + K \cos \beta)}{2}$$

if losses in nozzle are neglected ;  $K_{v_1} = 1$   
 Bucket friction is neglected (ideal flow) ;  $K = 1$

$$(\eta_{h, \max}) = 1 + \frac{\cos \beta}{2} \quad \beta \rightarrow \text{Blade angle}$$

$$\beta = 180 - \theta$$

$$\eta_{h, \max} = \frac{1 - \cos \theta}{2} \quad \theta \rightarrow \text{deflection Angle.}$$

if  $\theta = 180^\circ$

$$\eta_{h, \max} = 100\%$$

In practice, however the deflection is limited to about  $165^\circ$  so that the water leaving the bucket may not hit the following bucket

Wheel efficiency  $\eta_w$

$$\eta_w = \frac{\text{Runner power}}{\text{power input to wheel}}$$

$$\eta_w = \frac{\rho Q (V_{w1} + V_{w2}) u}{\frac{1}{2} \rho Q V_1^2}$$



### Nozzle efficiency $\eta_n$

$$\eta_n = \frac{\frac{1}{2} \rho Q V_1^2}{\rho g Q H} = \frac{V_1^2}{2gH} = \frac{K V_1^2 \cdot 2gH}{2gH} = K V_1^2$$

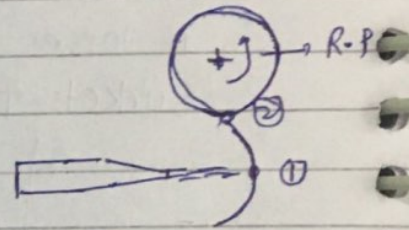
$$\eta_n = K V_1^2$$

$$\eta_h = \eta_n \times \eta_w \quad **$$

### Hydraulic losses

1. Nozzle losses =  $\rho g Q H - \frac{1}{2} \rho Q V_1^2$

2. Bucket friction losses =  $(h_{L, bf})$   
Applying energy eq<sup>n</sup> b/w ① & ②



$$\frac{V_1^2}{2g} = \frac{V_2^2}{2g} + h_{L, bf} + \frac{(V_{w1} + V_{w2}) u}{g}$$

$$\frac{V_{r1}^2}{2g} = \frac{V_{r2}^2}{2g} + h_{L, bf}$$

$$h_{L, bf} = \frac{V_{r1}^2 - V_{r2}^2}{2g}$$

3. K.E of water leaving the bucket

### Mechanical losses

$$S.P = R.P - \text{Mechanical losses}$$

- Bearing friction loss
- Windage loss.



### Working proportion of pelton wheel.

(i)  $V_1 = K_{V_1} \sqrt{2gH}$

$K_{V_1}$  = COEFF. of Velocity of jet

$$K_{V_1} = 0.97 - 0.99$$

(ii)  $U = K_u \sqrt{2gH}$

$K_u$  → speed ratio

$$K_u = 0.43 - 0.47$$

(iii)  $Q = \frac{\pi}{4} d^2 \times V_1$  ; for single jet

$$Q = n \times \frac{\pi}{4} d^2 \times V_1 ; \text{ for } n \text{ jets}$$

(iv) jet ratio (m)

$$m = \frac{D}{d} \rightarrow \begin{array}{l} D \rightarrow \text{Pitch circle dia of wheel} \\ d \rightarrow \text{Dia of jet} \end{array}$$

(v) Number of bucket (Z)

$$Z = 0.5m + 15 \quad m \rightarrow \text{jet ratio}$$

(vi) Selection of Wheel

$$U = K_u \sqrt{2gH}$$

$$\frac{\pi D N}{60} = K_u \sqrt{2gH}$$

$$\Rightarrow N = \frac{60 K_u \sqrt{2gH}}{\pi D}$$

Also, speed of generator is calculated as -

$$N = \frac{60 \times f}{P}$$

f; frequency (= 50 Hz in India)

P: no. of pair of poles

$$N = \frac{120 \times f}{P'}$$

P' → No. of poles.

N → synchronous speed.



## Governing of Turbine

Governer regulates the inflow of water into a turbine. A hydro turbine is directly coupled to the electric generator hence is req<sup>d</sup> to maintain synchronous speed @ all loads. Change in load necessitates speed adjustment.

The req<sup>d</sup> adjustments in form of control of flow into the turbine is done by governer. in pelton wheel governing is done through nozzle and sphere arrangement by moving the sphere (Needle Valve)

### Runaway speed

When discharge is hitting the bucket and load suddenly reduces to 0, wheel rotates @ max speed and this speed is known as Runaway speed.

### Specific speed ( $N_s$ )

$$N_s = \frac{N\sqrt{P}}{H^{5/4}}$$

$N \rightarrow$  speed of wheel (rpm)

$P \rightarrow$  power output for a single jet (kW)

$H \rightarrow$  Net Head (m)

for  $n$ -no. of jets

$$Q_{n\text{-jet}} = n \times \frac{\pi}{4} \times d^2 \times V_i$$

$$P_{n\text{-jet}} = n \times P_{\text{single jet}}$$

$$(N_s)_{n\text{-jet}} = \frac{N\sqrt{n P_{\text{single jet}}}}{(H)^{5/4}}$$

$$= \frac{\sqrt{n} N\sqrt{P}}{H^{5/4}} = \sqrt{n} (N_s)_{\text{single jet}}$$



Que. In a 2 jet pelton wheel working @ full load, each of the nozzles are issuing 10 cm dia jets. The nozzles are so shaped so as to avoid any further contraction of jet. if the load is suddenly reduced to 36% of full load, the altered jet dia will be?

A) 1.8 cm

B) 3 cm

C) 3.6 cm

D) 6 cm.

$$Q = n \times \frac{\pi}{4} d^2 \times V_1$$

$$Q \propto d^2$$

As load is reduced to 36% of full load

$$\frac{Q_2}{Q_1} = 0.36$$

$$\frac{Q_2}{Q_1} \propto \left(\frac{d_2}{d_1}\right)^2$$

$$0.36 = \left(\frac{d_2}{10}\right)^2 \Rightarrow \boxed{d_2 = 6 \text{ cm}}$$

Que. Following observations were obtained from test

Head @ Base of nozzle = 32 m

Discharge of nozzle =  $0.18 \text{ m}^3/\text{s}$

Area of jet =  $7500 \text{ mm}^2$

Power available @ shaft = 44 kW

Mechanical efficiency = 94%

Calculate power lost in kW in ~~first~~ i) Nozzle

ii) Runner iii) mechanical friction

$$V = \frac{Q}{A} = \frac{0.18}{0.0075} = 24 \text{ m/s}$$

$$R.P = \frac{44}{\eta_m} = \frac{44}{0.94} = 46.80 \text{ kW}$$



$$i) \text{ power lost in nozzle} = \rho Q H - \frac{1}{2} \rho Q V_1^2$$

$$= 1000 \times 9.81 \times 0.18 \times 32 - \frac{1}{2} \times 1000 \times 0.18 \times 24^2$$

$$= 4.67 \text{ KW}$$

$$ii) \text{ power lost in runner} = \frac{1}{2} \rho Q V_1^2 - R.P$$

$$= \frac{1}{2} \times 1000 \times 0.18 \times 24^2 - \frac{44}{0.94}$$

$$= 5.03 \text{ KW}$$

$$iii) \text{ power lost in Mechanical friction} = R.P - S.P$$

$$= \frac{44}{0.94} - 44$$

$$= 2.808 \text{ KW}$$

Que pelton Wheel is producing a Shaft power of 1 MW while working under a net head of 150 m. speed  $N = 450 \text{ RPM}$ ,  $K_{v_1} = 0.97$

$$K = 0.9, \theta \text{ deflection angle} = 165^\circ \quad \beta = 180 - 165$$

$$\text{Mechanical loss} = 5\% \text{ of Velocity head} = 15\%$$

$$\text{of jet pitch circle Dia of runner} = 1.2 \text{ m}$$

calculate i) mechanical losses expressed in m of water ii) Hydraulic losses in m of water ~~indic~~ indicating all components separately iii) Hydraulic efficiency, mechanical efficiency + overall efficiency of turbine.

$$V_1 = K_{v_1} \sqrt{2gH}$$

$$V_1 = 0.97 \sqrt{2 \times 9.81 \times 150}$$

$$V_1 = 52.62 \text{ m/s}$$

$$\text{Mechanical loss} = 0.05 \times \frac{V_1^2}{2g} = 7.05 \text{ m}$$



## Hydraulic losses

$$\text{i) nozzle loss} = H - \frac{V_1^2}{2g} = 150 - \frac{(52.62)^2}{2 \times 9.81} = 8.87 \text{ m}$$

## ii) Bucket friction loss

$$h_{L, bf} = \frac{V_{r1}^2 - V_{r2}^2}{2g}$$

$$V_{r1} = V_1 - u$$

$$V_{r1} = 24.36 \text{ m/s}$$

$$V_{r2} = K V_{r1}$$

$$V_{r2} = 0.9 \times 24.36$$

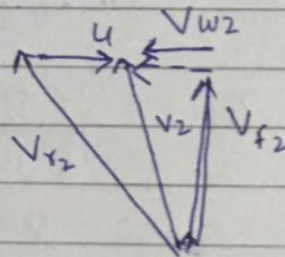
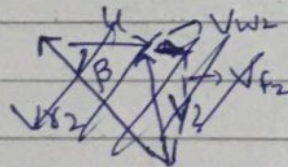
$$V_{r2} = 21.924 \text{ m/s}$$

$$\left\{ \begin{aligned} u &= \frac{\pi D N}{60} \\ &= \frac{3.14 \times 1.2 \times 450}{60} \\ u &= 28.26 \end{aligned} \right.$$

$$h_{L, bf} = \frac{24.36^2 - 21.924^2}{2 \times 9.81}$$

$$h_{L, bf} = 5.74 \text{ m}$$

## iii) KE of Water leaving the bucket = $\frac{V_2^2}{2g}$



$$V_2^2 = V_{w2}^2 + V_{f2}^2$$

$$V_{w2} = V_{r2} \cos \beta - u$$

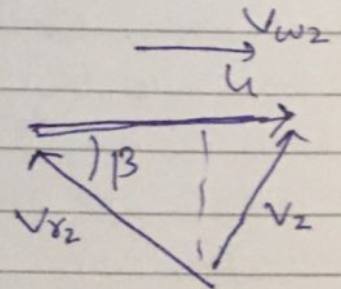
$$= 21.92 \cos 15^\circ - 28.26$$

$$= -7.086 \text{ m/s}$$

$$V_{f2} = V_{r2} \sin \beta$$

$$= 21.92 \sin 15^\circ$$

$$= 5.6 \text{ m/s}$$





$$V_2^2 = V_{w2}^2 + V_{t2}^2$$

$$= (7.086)^2 + (5.6)^2$$

$$V_2 = 9.03 \text{ m/s}$$

KE of Water Leaving the Bucket

$$\frac{V_2^2}{2g} = \frac{(9.03)^2}{2 \times 9.81} = 4.16 \text{ m}$$

Total hydraulic losses,

$$= 8.875 + 5.74 + 4.16$$

$$= 18.81 \text{ m}$$

$$\text{iii) } \eta_h = \frac{R.P}{W.P} = \frac{H_e}{H}$$

$$R.P (H_e) = ~~W.P (H)~~ H - \text{hydraulic losses.}$$

$$= 150 - 18.81$$

$$= 131.19 \text{ m}$$

$$\eta_h = \frac{131.19}{150} = 0.8746$$

$$\eta_m = \frac{S.P}{R.P}$$

$$S.P = R.P - \text{mechanical losses.}$$

$$= 131.19 - 7.057$$

$$= 124.14 \text{ m}$$

$$\eta_m = \frac{124.14}{131.18} = 0.946$$

$$\eta_o = \eta_m \times \eta_h = 0.946 \times 0.875$$

$$= 82.7 \%$$



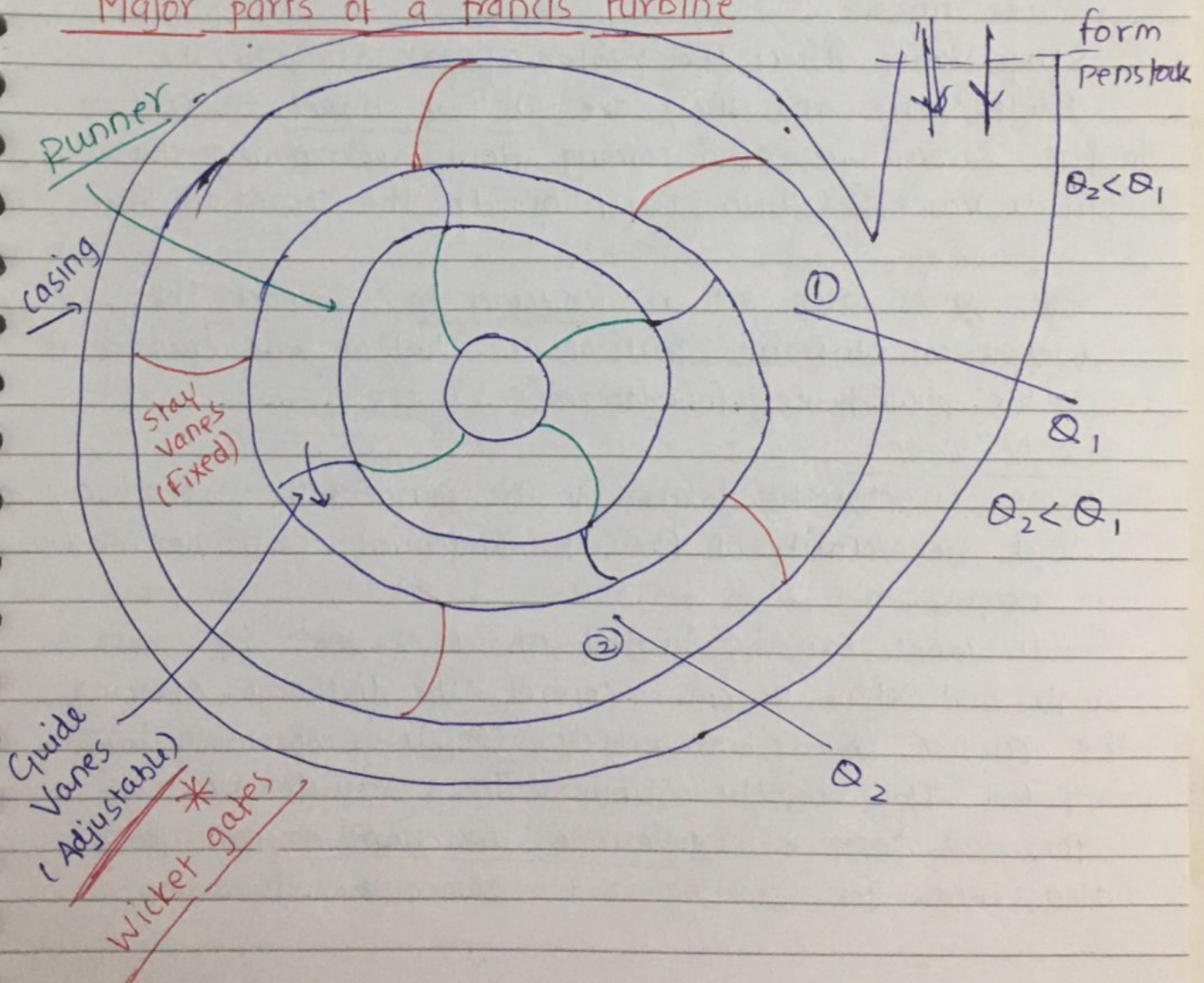
## Reaction Turbine

In this energy at the entrance to runner is in form of both K.E and P.E, while passing through runner passage both pressure & velocity ↓ and @ exit pressure is -ve and velocity is low.

In reaction turbines runner blades are completely submerged in water

In reac<sup>n</sup> turbines the flow entering the casing is turned towards the runner by the stay vanes and then it passes through guide vanes and runner blades.

### Major parts of a francis turbine





casing It is a closed passage of gradually reducing c/s enclosing the Runner. Its function is to distribute the flow over the periphery of the runner in such a way that the entrance velocity to the runner vanes is same all around it.

Since the flow is pressurized flow, casing should be an air-tight casing.

### Stay Vanes

These are optional and are generally provided for large turbines. and these are fixed vanes outside the Guide vanes.

Stay vanes direct the water from casing to the guide vane and they are air foil shaped to conform to the streamlines of incoming flow approaching the Guide Vane which will avoid the separation losses.

Stay vanes also act as column to support the weight of different parts of the turbine and transfer it to the powerhouse foundation.

### Guide Vanes

These direct the water to the runner in such a way that the velocity of flow is tangential to runner blade @ entry.

Guide vanes can be turned about the pivot by small angle and thus it can control the discharge entering the runner. This also includes complete shutting off of the flow to the runner. Thus Guide Vanes form a part of governing mechanism and are also used for starting and stopping the turbine.



#### iv) Runner

Rotating component of turbine. When water is allowed to flow through the runner, its angular momentum is reduced, thereby producing the torque, which rotates the runner. From the runner flow is directed to a draft tube and it is always desirable that water leaves the runner with no angular momentum.

#### DRAFT TUBE

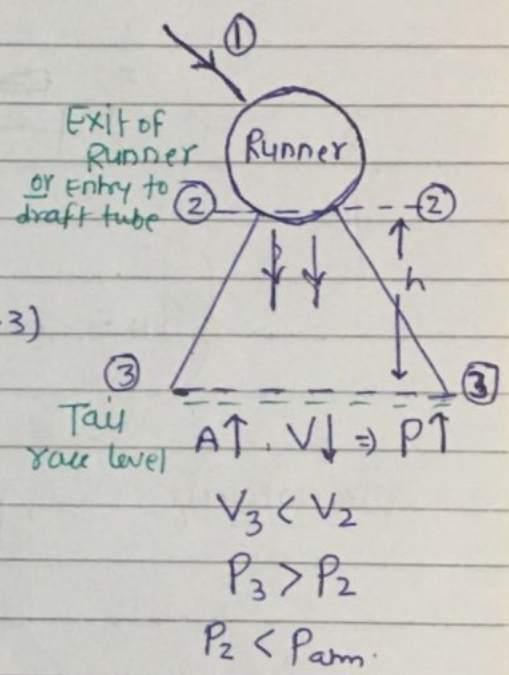
KE is significant

$$R.P = \left( \frac{P_1 - P_2}{\rho g} \right) + \left( \frac{V_1^2 - V_2^2}{2g} \right)$$

Applying Energy Eq<sup>n</sup> b/w sec<sup>n</sup> (2-2) & (3-3)

$$\frac{P_2}{\rho g} + \frac{V_2^2}{2g} + h = \frac{P_3}{\rho g} + \frac{V_3^2}{2g} + h_{Ldt}$$

$$\boxed{\frac{P_2}{\rho g} = -h - \frac{V_2^2}{2g} + \frac{V_3^2}{2g} + h_{Ldt}}$$



In reaction turbines K.E @ exit of runner is significant. This K.E can be converted into pressure energy by gradual expansion of flow through a gradually diverging device known as draft tube.

Draft tube serves 2 purposes

- i) it converts K.E @ exit of runner into useful pressure energy
- ii) Turbine can be located above the tail race without compromising on available head.



## Efficiency of Draft tube

$$= \frac{\text{Actual conversion of kinetic head into pressure head}}{\text{Max. possible recovery of kinetic head}}$$

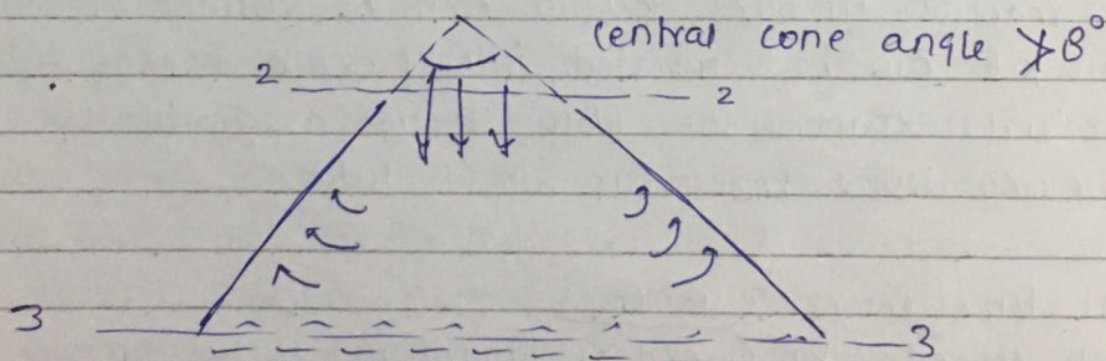
$$= \frac{\frac{V_2^2}{2g} - \frac{V_3^2}{2g} - h_{l,dt}}{\frac{V_2^2}{2g} - \frac{V_3^2}{2g}}$$

$$= 1 - \frac{\text{loss inside the Draft tube}}{\frac{V_2^2 - V_3^2}{2g}}$$

$$\eta_{dt} = 1 - \frac{h_{l,dt}}{\frac{V_2^2 - V_3^2}{2g}}$$

Alternatively,

$$\eta_{dt} = \frac{\text{Actual conversion of kinetic head into pressure head}}{\text{Kinetic head at entrance of Draft tube.}}$$



central cone angle of straight divergent draft tube should not be more than  $8^\circ$  otherwise flow separation may occur and losses in draft tube will be more.



The lower end of draft tube should be kept submerged in water so that air can't enter through it.

### Cavitation in Reaction Turbines

$P_{abs} < P_{vap.} \Rightarrow$  cavitation may occur

$$\left(\frac{P_2}{\rho g}\right)_{abs.} = \left(\frac{P_3}{\rho g}\right)_{abs.} - h - \frac{V_2^2}{2g} + \frac{V_3^2}{2g} + h_{L,dt}$$

$$\left(\frac{P_2}{\rho g}\right)_{abs.} = \frac{P_{atm}}{\rho g} - h - \frac{V_2^2}{2g} + \frac{V_3^2}{2g} + h_{L,dt}$$

for no cavitation,

$$\left(\frac{P_2}{\rho g}\right)_{abs.} > \left(\frac{P_v}{\rho g}\right)$$

$$\frac{P_{atm}}{\rho g} - h - \frac{V_2^2}{2g} + \frac{V_3^2}{2g} + h_{L,dt} > \frac{P_v}{\rho g} \quad \text{--- ①}$$

chances of cavitation are high if-

$$\begin{array}{l} P_{atm} \downarrow \\ h \uparrow \\ V_2 \uparrow \\ T \uparrow \Rightarrow P_v \uparrow \end{array}$$

from eq<sup>n</sup> ①  $h < \frac{P_{atm}}{\rho g} - \frac{P_v}{\rho g} - \frac{V_2^2}{2g} + \frac{V_3^2}{2g} + h_{L,dt}$

Max. height @ which turbine can be located above TRL

$$(h)_{max} = \frac{P_{atm} - P_v}{\rho g} - \frac{V_2^2 - V_3^2}{2g} + h_{L,dt}$$



Alternatively

check for cavitation can be done by using

$$\text{Thoma's cavitation factor } (\sigma) = \frac{H_a - H_s - H_v}{H}$$

$H_a \rightarrow$  Atmospheric pressure head

$H_s \rightarrow$  Draft head (height of turbine above TRL)

$H_v \rightarrow$  Vapor pressure head

$H \rightarrow$  net head

for no cavitation ;  $\sigma > \sigma_c$

$\sigma_c$  : critical cavitation factor

$$\sigma_c = f(N_s)$$

$N_s \uparrow \rightarrow \sigma_c \uparrow \rightarrow \text{chances of cavitation} \uparrow$

$\sigma$  is a <sup>measure</sup> ~~major~~ of margin against cavitation susceptibility

In reaction turbine pressure is Min. @ exit of runner or at the entry to draft tube hence chances of cavitation are max @ the exit of runner

Ques Giving that atmospheric pressure head = 9 m.

Vapor pressure head = 1 m, net head = ~~40~~ 40 m

Cavitation coefficient  $\sigma = 0.15$ , the height @ which

Turbine can be located above TRL is

- a) 6m b) 4m c) 3m d) 2m

Ques Francis turbine operating under a net head of 20 m is tentatively proposed to be set up with a draft head of 2 m @ a certain location.



local atmospheric head = 10 m, and vapor pressure head is 0.18 m. critical cavitation parameter for given turbine can be taken as 0.4

- i) estimate whether there is Cavitation susceptibility.
- ii) if a safety margin of 0.5 m for draft height is mandatory, what is the max height above tail water @ which turbine can be installed.

$$\sigma = \frac{10 - 2 - 0.18}{20}$$

$$\sigma = 0.391 < \sigma_c = 0.4$$

hence, cavitation susceptibility =

$$0.4 = \frac{10 - x - 0.18}{20}$$

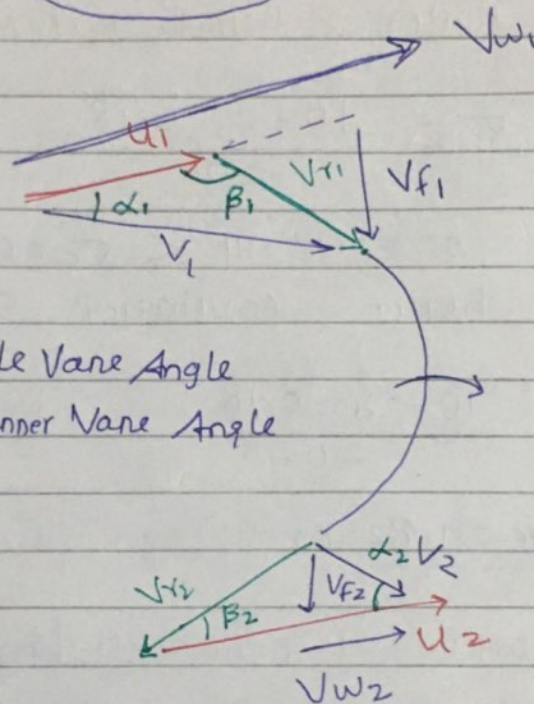
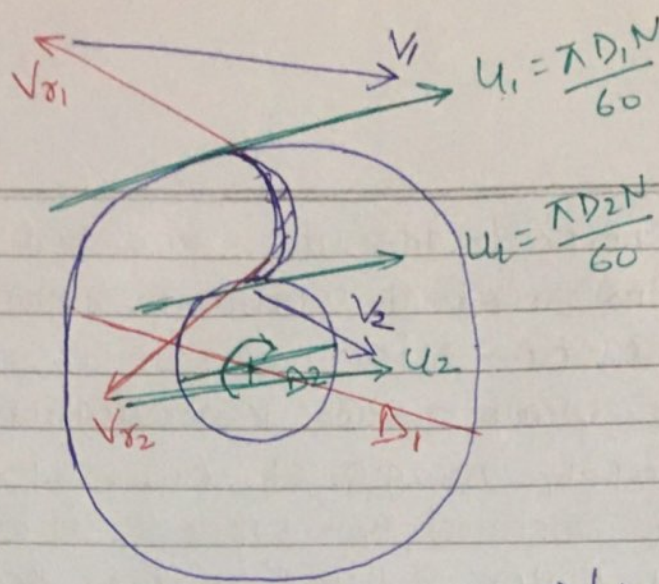
$$x = 1.82 \text{ m}$$

after margin of 0.5 m  $\Rightarrow (h_s)_{\max} = 1.32 \text{ m}$

Que Conical draft tube of a react turbine has dia of 1.5 m and 2 m at its ends the turbine is set 7 m above Tail level When discharge from turbine is  $12 \text{ m}^3/\text{sec}$  the loss of energy in draft tube is  $0.35 \times \text{Velocity head @ Runner exit take}$ . Atmospheric pressure head as 10.3 m of water calculate i) pressure head @ entrance of draft tube ii) efficiency of draft tube.



## RUNNER POWER



$V_1$  &  $U_1 \rightarrow \alpha_1 \rightarrow$  Guide Vane Angle  
 $V_{r1}$  &  $-U_1 \rightarrow \beta_1 \rightarrow$  Runner Vane Angle

- $U_1$  and  $U_2$  are peripheral or tangential velocity of runner @ inlet and exit
- $V_{r1}$  and  $V_{r2}$  are relative velocity of water @ inlet and exit
- NOTE For shockless entry and exit  $V_{r1}$  &  $V_{r2}$  should be tangential to the blade
- $\alpha_1$  and  $\alpha_2$  are Guide vane angle @ inlet & exit
- $\beta_1$  and  $\beta_2$  runner vane angle @ inlet & exit
- $V_{w1}$  and  $V_{w2}$  are tangential components of absolute velocity @ inlet and exit (Whirl velocity)
- $V_{f1}$  and  $V_{f2} \rightarrow$  radial component of absolute velocity @ inlet and exit. (Flow velocity)



moment of momentum principle is newton's 2<sup>nd</sup> law applied for a rotating fluid mass.

In a steady state flow the rate of change of moment of momentum of a system = net torque acting on the system.

$$\text{Runner power} = W \times T_{\text{water on runner}}$$

$$T_{\text{water on runner}} = - T_{\text{runner on water}}$$

$$T_{\text{runner on water}} = \text{Rate of change of angular momentum of water in c/w dir}^n$$

$$= \text{Angular momentum flux out in } \curvearrowright \text{ dir}^n - \text{Angular momentum flux in } \curvearrowright \text{ dir}^n$$

$$= \rho Q V_{w2} r_2 - \rho Q V_{w1} r_1$$

$$T_{\text{water on runner}} = - [\rho Q V_{w2} r_2 - \rho Q V_{w1} r_1]$$
$$= \rho Q [V_{w1} r_1 - V_{w2} r_2]$$

$$\text{Runner power} = W \times T_{\text{water on runner}}$$
$$= \rho Q [V_{w1} r_1 - V_{w2} r_2] \omega$$
$$= \rho Q [V_{w1} (\omega r_1) - V_{w2} (\omega r_2)]$$

$$R.P = \rho Q [V_{w1} u_1 - V_{w2} u_2] \quad \alpha_2 < 90^\circ$$

$$R.P = \rho Q [V_{w1} u_1 + V_{w2} u_2] \quad \alpha_2 > 90^\circ$$

if  $\alpha_2 = 90^\circ$  Radial discharge  $V_{w2} = 0$   
@ outlet

$$V_2 = V_{f2}$$



practically it is observed that hydraulic efficiency of francis turbine is max. when discharge is radial @ the outlet, i.e. ~~water~~  $\alpha_2 = 90^\circ$ ,  $V_{w2} = 0$ ,  $V_2 = V_{f2}$   
 मतलब Water has no angular momentum.

$$\underline{R.P = \rho Q V_{w1} u_1}$$

Euler's Head ( $H_e$ )

$$= \frac{\text{Runner power}}{\text{Wt. of fluid}}$$

$$H_e = \frac{\rho Q (V_{w1} u_1 - V_{w2} u_2)}{\rho Q g} = \frac{V_{w1} u_1 - V_{w2} u_2}{g}$$

if  $\alpha_2 = 90^\circ$

$$H_e = \frac{V_{w1} u_1}{g}$$

for max  $\eta_h$

for old francis

Discharge is purely radial @ outlet }  
Modern francis Discharge is purely axial @ outlet }

Radial Discharge @ outlet

$$\alpha_2 = 90^\circ, V_{w2} = 0$$

$$V_2 = V_{f2}$$

\*\*\*

Degree of Reaction in Reaction Turbines

$$R.P = \left( \frac{P_1 - P_2}{\rho g} \right) + \left( \frac{V_1^2 - V_2^2}{2g} \right)$$

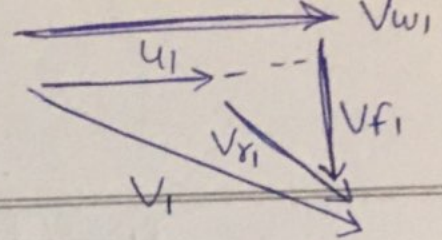
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Reaction effect

↓

Impulse effect





$$H_e = \frac{V_{w1} u_1 - V_{w2} u_2}{g}$$

inlet  
from velocity triangle.

$$V_1^2 = V_{w1}^2 + V_{f1}^2$$

$$V_1^2 = V_{w1}^2 + V_{r1}^2 - (V_{w1} - u_1)^2$$

$$V_1^2 = V_{w1}^2 + V_{r1}^2 - V_{w1}^2 - u_1^2 + 2u_1 V_{w1}$$

$$V_{w1} u_1 = \frac{V_1^2 + u_1^2 - V_{r1}^2}{2}$$

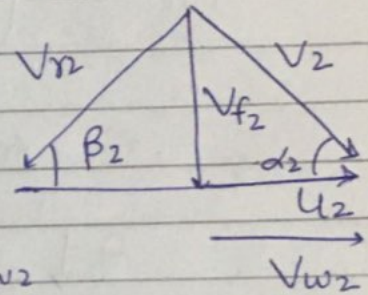
from exit velocity triangle

$$V_2^2 = V_{w2}^2 + V_{f2}^2$$

$$V_2^2 = V_{w2}^2 + V_{r2}^2 + (u_2 - V_{w2})^2$$

$$V_2^2 = V_{w2}^2 + V_{r2}^2 + u_2^2 + V_{w2}^2 + 2u_2 V_{w2}$$

$$V_{w2} u_2 = \frac{V_2^2 + u_2^2 - V_{r2}^2}{2}$$

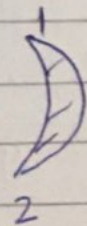


$$H_e = \left( \frac{V_1^2 + u_1^2 - V_{r1}^2}{2g} \right) - \left( \frac{V_2^2 + u_2^2 - V_{r2}^2}{2g} \right)$$

$$H_e = \left( \frac{V_1^2 - V_2^2}{2g} \right) + \left( \frac{u_1^2 - u_2^2}{2g} \right) + \left( \frac{V_{r2}^2 - V_{r1}^2}{2g} \right) \quad \text{--- (1)}$$

Applying Energy Eq<sup>n</sup> b/w inlet and exit of runner  
(Neglecting losses in runner)

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + Z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + Z_2 + H_e$$





$$H_e = \left( \frac{P_1 - P_2}{\rho g} \right) + \left( \frac{V_1^2 - V_2^2}{2g} \right) \quad \text{--- (2)}$$

From eq<sup>n</sup> ① & ②

$$\left( \frac{P_1 - P_2}{\rho g} \right) + \left( \frac{V_1^2 - V_2^2}{2g} \right) = \left( \frac{V_1^2 - V_2^2}{2g} \right) + \left( \frac{u_1^2 - u_2^2}{2g} \right) + \left( \frac{V_{r2}^2 - V_{r1}^2}{2g} \right)$$

$$\left( \frac{P_1 - P_2}{\rho g} \right) = \left( \frac{u_1^2 - u_2^2}{2g} \right) + \left( \frac{V_{r2}^2 - V_{r1}^2}{2g} \right) \quad \text{--- (3)}$$

$$H_e = \left( \frac{V_1^2 - V_2^2}{2g} \right) + \left( \frac{u_1^2 - u_2^2}{2g} \right) + \left( \frac{V_{r2}^2 - V_{r1}^2}{2g} \right)$$

↓  
Impulse effect

↓  
pressure effect

$$A \downarrow \Rightarrow V_r \uparrow \Rightarrow P \downarrow$$

Degree of Reaction  $\rightarrow R = \frac{\frac{P_1 - P_2}{\rho g}}{H_e}$

impulse turbine  $\rightarrow P_1 = P_2 = P_{atm}$

Degree of reaction = 0

\* ~~Kaplan~~ Kaplan and Francis & propeller turbine are not pure (100%) reaction turbines.

•  $\frac{V_1^2 - V_2^2}{2g}$  represents decrease in K.E of fluid

•  $\frac{u_1^2 - u_2^2}{2g}$  represents change in energy in terms of pressure head due to centrifugal action

•  $\frac{V_{r2}^2 - V_{r1}^2}{2g}$  represents change in energy in terms of pressure head due to acceleration of fluid while flowing through the runner passage.



## Efficiency of Reaction Turbine

### ① Hydraulic efficiency

$$\eta_h = \frac{R.P}{W.P} = \frac{\rho Q V_{w1} u_1}{\rho Q g H}$$

$$\eta_h = \frac{V_{w1} u_1}{g H}$$
$$\eta_h = \frac{H_e}{H(\text{net})}$$

### ② Mechanical efficiency

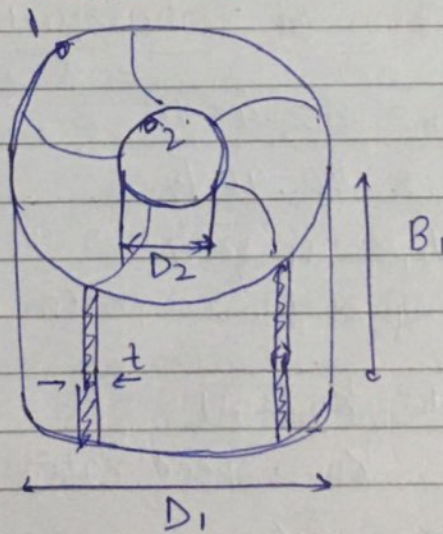
$$\eta_m = \frac{S.P}{R.P}$$

### ③ Overall Efficiency

$$\eta_o = \eta_h \eta_m$$

$$\eta_o = \frac{S.P}{W.P}$$

## Discharge through the turbine



$D_1$  &  $D_2 \rightarrow$  Dia at inlet & exit

$B_1$  &  $B_2 \rightarrow$  width of runner @ inlet & exit

$$Q = \pi D_1 B_1 \times V_{f1}$$

in  $n \rightarrow$  no. of Vanes

$t \rightarrow$  thickness of Vane

$$Q = (\pi D_1 - nt) B_1 \times V_{f1} = K \pi D_1 B_1 V_{f1}$$

$K \rightarrow$  Blade thickness Coeff. = 0.95



For old Francis turbine

$$\text{Radial entry } Q = \pi D_1 B_1 \times V_{f1}$$

$$\text{Radial exit } Q = \pi D_2 B_2 \times V_{f2}$$

Modern Francis turbine

$$\text{Radial entry } Q = \pi D_1 B_1 \times V_{f1}$$

$$\text{Axial exit } Q = \frac{\pi D_2^2}{4} \times V_{f2}$$

Unless otherwise mentioned it is assumed that

i) Water leaves the runner with no angular momentum, or discharge is radial @ the outlet i.e.  $\alpha_2 = 90^\circ$

$$V_{w2} = 0, V_2 = V_{f2}$$

ii) Flow velocity  $V_f$  is constant i.e.  $V_{f1} = V_{f2}$

Working proportion of Francis turbine

$$1) \eta_h = 85-95\%$$

$$2) \eta_o = 80-90\%$$

$$3) V_{f1} = \psi \sqrt{2gH}$$

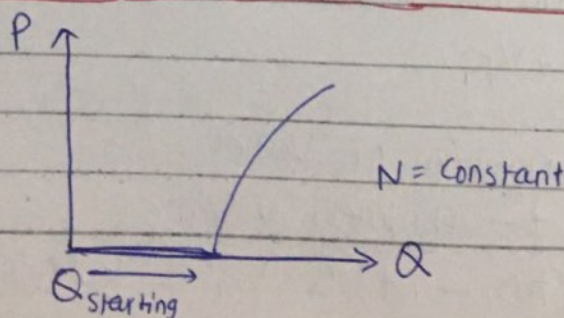
$$\psi \rightarrow \text{flow ratio} = 0.15-0.3$$

$$4) U_1 = K_u \sqrt{2gH}$$

$$K_u \rightarrow \text{speed ratio} = 0.6-0.9$$

$$5) \eta_o = B_1/D_1 = 0.1-0.45$$

Operating characteristics of Francis turbine





- A Turbine is Used by coupling it to generator for power generator and hence its operating speed is constant
- A min. Discharge,  $Q_{\text{starting}}$  is Required to overcome the inertia of turbine unit, which requires a min. Torque to overcome it before it starts moving from the ~~idle~~<sup>idle</sup> position.

### For PRE

- 1) PE & KE Both
- 2) Runner blades are Completely Submerged in Water
- 3) Kaplan  $\rightarrow$  Axial  
modern  $\rightarrow$  mixed  
old Francis  $\rightarrow$  Radial
- 4)  $N_s \rightarrow 60-300$   
 $300-900$
- 5) casing  $\rightarrow$  Air tight  
cls is gradually reducing to keep entering Velocity same @ all blades
- 6) Stay Vanes  $\rightarrow$  Fixed  
Airfoil shape  
behaves @ column.
- 7) Guide Vanes  $\rightarrow$  Adjustable  
form a part of governing mechanism
- 8) Runner  $\rightarrow$
- 9) Draft tube  $\rightarrow$

$$10) \eta_{dt} = \frac{\frac{V_2^2}{2g} - \frac{V_3^2}{2g} - h_{\text{exit}}}{\frac{V_2^2}{2g} - \frac{V_3^2}{2g}}$$

Draft tube separation angle  $\neq 8^\circ$



(11) cavitation chances

for no  
cavitation

$$\left(\frac{P_2}{\rho g}\right)_{\text{abs}} > \left(\frac{P_{\text{vap}}}{\rho g}\right)$$

$$h_{\text{max.}} = \frac{P_{\text{atm}} - P_v}{\rho g} - \frac{V_2^2 - V_3^2}{2g} + h_{\text{c,d}} + h_{\text{f,d}}$$

Also  $\sigma > \sigma_c$

$$\sigma = \frac{H_a - H_s - H_v}{H}$$

(12)  $Q = \pi D_1 B_1 V_{f1}$

(13)  $Q_{\text{starting}}$

Ques A Francis turbine has dia of 1.5 m. and rotates at 450 rpm, velocity of flow is 10 m/s and discharge is radial @ outlet, Turbine develops 13 mw of power with a discharge of 12 cumec and diff<sup>n</sup> of piezometric head b/w the entrance and exit from the runner is 70 m. Find

i) Absolute velocity and guide Vane angle @ inlet

ii) Angle of Runner blade @ entry.

iii) loss of head in the runner.

$$D_1 = 1.5 \text{ m}$$

$$N = 450 \text{ rpm}$$

$$V_{f1} = 10 \text{ m/s} = V_{f2} = V_2$$

$$R.P = 13 \text{ mw}$$

$$Q = 12 \text{ m}^3/\text{s}$$



$$\left( \frac{P_1}{\rho g} + z_1 \right) - \left( \frac{P_2}{\rho g} + z_2 \right) = 70 \text{ m}$$

$$(i) \quad u_1 = \frac{\pi D_1 N}{60}$$

$$= \frac{\pi \times 1.5 \times 450}{60} = 35.34 \text{ m/s}$$

$$V_1 = \sqrt{V_{w1}^2 + V_{f1}^2}$$

$$R \cdot P = P Q V_{w1} u_1 = 13 \times 10^6 \text{ W}$$

$$\Rightarrow 1000 \times 12 \times V_{w1} \times 35.34 = 13 \times 10^6$$

$$V_{w1} = 30.65 \text{ m/s}$$

$$V_1 = 32.24 \text{ m/s}$$

$$\tan \alpha_1 = \frac{V_{f1}}{V_{w1}} = \frac{10}{30.65}$$

$$\alpha_1 = 18.06^\circ$$

$$(ii) \quad \tan \beta_1 = \frac{V_{f1}}{u_1 - V_{w1}} = \frac{10}{35.34 - 30.65} \Rightarrow \beta_1 = 64.85^\circ$$

$$(iii) \quad h_{L, \text{runner}}$$

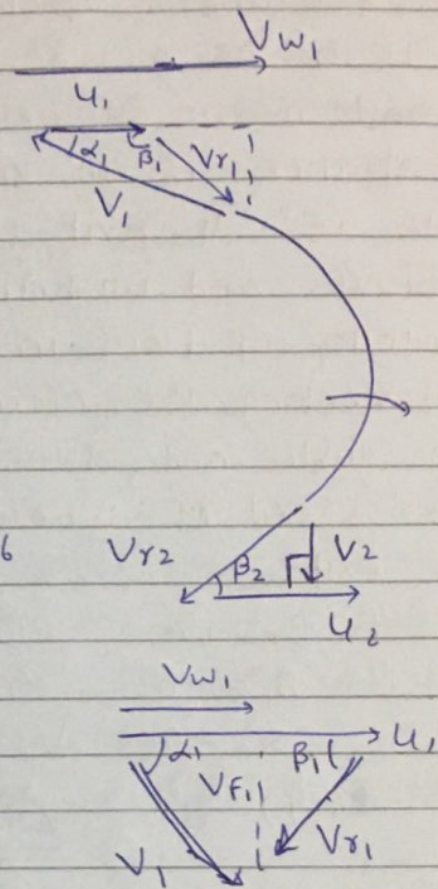
by energy eq<sup>n</sup>.

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + z_2 + h_{L, \text{runner}} + \frac{V_{w1} u_1}{g}$$

$$70 + \frac{V_1^2}{2g} - \frac{V_2^2}{2g} - \frac{V_{w1} u_1}{g} = h_{L, \text{runner}}$$

$$70 + \frac{32.24^2}{2 \times 9.8} - \frac{10^2}{2 \times 9.8} - \frac{30.65 \times 35.34}{9.8} = h_{L, \text{runner}}$$

$$\boxed{h_{L, \text{runner}} = 7.47 \text{ m}}$$





Ques A Francis turbine working under a head of 40 m. as a wheel dia of 1 m. @ the entrance and 0.5 m @ exit. The runner vane angle @ entrance is  $90^\circ$  and guide blade angle @ inlet is  $15^\circ$ . At exit water leaves the vane without shock and without any tangential velocity. Velocity of flow in runner is constant, Neglecting the effect of Draft tube and the losses in guide and runner passage, determine the speed of wheel in rpm and the vane angle @ exit.

$$D_1 = 1 \text{ m} \quad D_2 = 0.5 \text{ m} \quad H = 40 \text{ m}$$

$$\beta_1 = 90^\circ, \quad \alpha_1 = 15^\circ$$

$$\alpha_2 = 90^\circ, \quad V_1 = V_{f2} \quad V_{w2} = 0$$

$$i) \quad H = \frac{V_{w1} u_1}{g} + \frac{V_2^2}{2g}$$

$$V_2 = V_{f2} = V_{f1}$$

$$\tan \alpha_1 = \frac{V_{f1}}{V_{w1}} = \frac{V_{f1}}{u_1}$$

$$V_{f1} = u_1 \tan \alpha_1$$

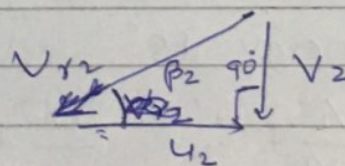
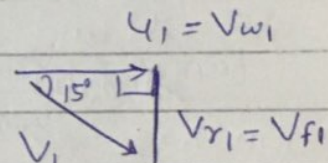
$$H = \frac{u_1^2}{g} + \frac{(u_1 \tan 15^\circ)^2}{2g}$$

$$u_1 = 19.46 \text{ m/s}$$

$$u_1 = \frac{\pi D_1 N}{60}$$

$$19.46 = \frac{3.14 \times 1 \times N}{60}$$

$$N = 371.7 \text{ rpm}$$





\* Kaplan turbine operates @ higher speed and hence its runner is smaller in size as compare to that of a Francis turbine

$$\tan \beta_2 = \frac{V_{f2}}{u_2}$$

$$u_2 = \frac{\pi D_2 N}{60}$$

$$\tan \beta_2 = \frac{V_{f1}}{9.72}$$

$$u_1 = 9.72 \text{ m/s}$$

$$\tan \beta_2 = \frac{u_1 \tan 15^\circ}{9.72}$$

$$\beta_2 = 28.18^\circ$$

### Kaplan and Propeller turbines

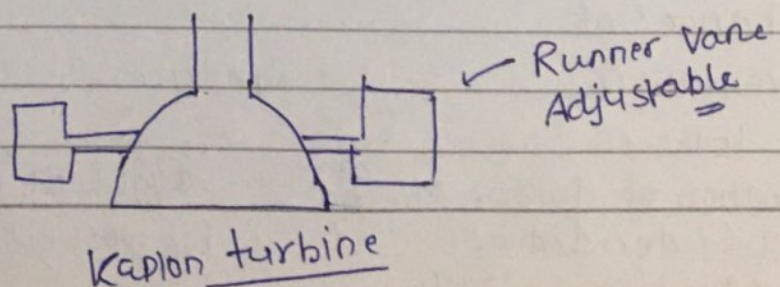
Both Kaplan & propeller turbines are axial flow turbines and the runner Vanes are connected to the boss or hub

The runner vane of propeller turbine are fixed while that of Kaplan turbine can be turned and hence part load efficiency of Kaplan turbine is higher as compare to that of a propeller turbine.

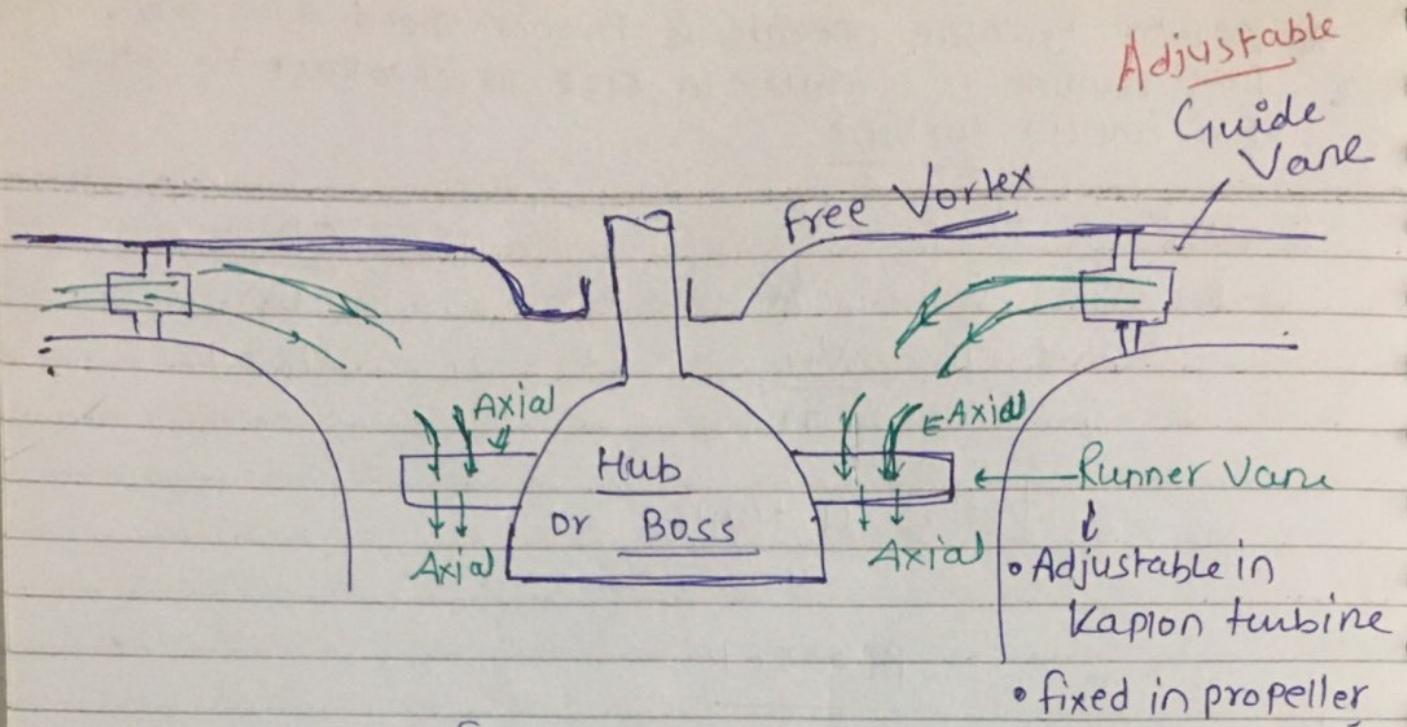
Francis turbine has 16-24 no. of blades while Kaplan and propeller turbine have 4-6 only. hence there is lesser contact surface area in Kaplan and propeller turbine therefore frictional losses are lesser in Kaplan and propeller turbine

space b/w the guide vane outlet and runner is known as Whirl chamber and in this chamber flow changes from radial to axial

- Reaction turbine
- Axial Flow turbine
- Kaplan turbine  $\rightarrow$  runner ~~blades~~ <sup>vane</sup> are adjustable.







Free Vortex  $\Rightarrow V_w \cdot r = \text{constant}$

\*

Pelton

Tangential

Impulse

High H, low Q

$N_s$  is low (8-30)

part load  $\eta$  is high

$P_1 = P_2 = P_{atm}$

• Based on fluctuations in TRL

Francis

Mixed / Radial

Reaction

Medium H,

Medium Q

~~40-450~~ 60-300

$P_1 > P_{atm}$

$P_2 < P_{atm}$

Location of turbine is decided by dT

Draft tube is always kept submerged in water so that air cannot enter

Kaplan / propeller

Axial

Reaction

low H

High Q

$N_s = 300-900$

part load  $\eta$  is max

Kaplan

propeller  
↓  
poor

chances of

cavitation

low

location of turbine above TRL is decided by fluctuation in TRL

medium

Max.

height of draft tube and hence by possibility of cavitation



## Model Analysis

To predict the behaviour of a turbine under Varying Cond. Unit quantities are defined

To estimate unit quantities it is assumed that overall efficiency is constant and for this velocity triangle under working head and other heads should be similar.

Such turbines are called homologous turbines

$$\frac{u_1}{u_2} = \frac{V_1}{V_2} = \frac{V_{r1}}{V_{r2}} = \frac{V_{w1}}{V_{w2}} = \frac{V_{f1}}{V_{f2}}$$

### Unit speed ( $N_u$ )

It is the theoretical speed @ which turbine would run while operating under unit head

$$100 \text{ m head} \rightarrow N_1$$

$$1 \text{ m head} \rightarrow N_u$$

$$u = \frac{\pi D N}{60}$$

$$u \propto D N$$

$$\text{Also, } u = K_u \sqrt{2gH}$$

$$u \propto \sqrt{H}$$

$$\text{so, } \sqrt{H} \propto D N$$

$$\frac{\sqrt{H}}{D N} = \text{constant} \Rightarrow$$

$$\frac{H}{D^2 N^2} = \text{constant}$$

for same turbine

$D$  is constant

$$\sqrt{H} \propto N$$

$$\frac{N_2}{N_1} = \sqrt{\frac{H_2}{H_1}}$$

$$H_1 = H ; N_1 = 1$$

$$H_2 = 1 ; N_2 = N_u$$

$$\Rightarrow \frac{N_u}{N} = \sqrt{\frac{1}{H}}$$

$$\Rightarrow N_u = \frac{N}{\sqrt{H}}$$



### Unit Discharge ( $Q_u$ )

Discharge passing through a turbine operating under unit head

$$Q = \frac{\pi}{4} D^2 \times V_f$$

$$Q \propto D^2 \cdot V_f$$

$$Q \propto D^2 \sqrt{H}$$

$$\{ V_f = \psi \sqrt{2gH}$$

$$\boxed{\frac{Q}{D^2 \sqrt{H}} = \text{constant}} *$$

For same turbine  $D = \text{constant}$

$$\frac{Q}{\sqrt{H}} = \text{const.}$$

$$\frac{Q_2}{Q_1} = \sqrt{\frac{H_2}{H_1}}$$

$$H_1 = H, \quad Q_1 = Q$$

$$H_2 = 1 \text{ m}, \quad Q_2 = Q_u$$

$$\frac{Q_u}{Q} = \sqrt{\frac{1}{H}}$$

$$\boxed{Q_u = \frac{Q}{\sqrt{H}}} \Rightarrow \frac{Q}{D^2 \sqrt{H}} = \text{constant}$$

$$\sqrt{H} \propto DN$$

$$\boxed{\frac{Q}{D^3 N} = \text{constant}} *$$

### Unit power ( $P_u$ )

$$P = \eta_o \rho g Q H$$

$$P \propto Q H$$

$$P \propto D^2 H^{3/2}$$

$$P \propto D^2 \sqrt{H} \cdot H \quad (Q \propto D^2 \sqrt{H})$$



$$\frac{P}{D^2 H^{3/2}} = \text{constant} *$$

for same turbine  $D \rightarrow \text{const.} \Rightarrow \frac{P}{H^{3/2}} = \text{const}$

$$P \propto H^{3/2}$$

$$\frac{P_2}{P_1} = \left( \frac{H_2}{H_1} \right)^{3/2}$$

$$H_1 = H ; P_1 = P$$

$$H_2 = 1 \text{ m} ; P_2 = P_u$$

$$\frac{P_u}{P} = \left( \frac{1}{H} \right)^{3/2}$$

$$\Rightarrow P_u = \frac{P}{H^{3/2}} *$$

Now,  $\frac{P}{D^2 H^{3/2}} = \text{constant}$

$$\sqrt{H} \propto DN$$

$$\frac{P}{D^2 (DN)^{3/2}} = \text{constant}$$

$$\boxed{\frac{P}{D^5 N^3} = \text{constant}}$$

for same turbine

$$\frac{H}{D^2 N^2} = \text{const}$$

$$\frac{\sqrt{H}}{DN} = \text{const}$$

$$N \propto \sqrt{H}$$

$$\frac{Q}{D^3 N} = \text{const}$$

$$\frac{Q}{D^2 \sqrt{H}} = \text{const.}$$

$$Q \propto \sqrt{H}$$

$$\frac{P}{D^5 N^3} = \text{const}$$

$$\frac{P}{D^2 (H)^{3/2}} = \text{const.}$$

$$P \propto H^{3/2}$$



## Specific Speed ( $N_s$ )

It is the rotational speed of a Homologous turbine so reduced inside that it produces 1 kW power working under a head of 1m. Under conditions of Max. efficiency.

$$\frac{P}{D^2 H^{3/2}} = \text{const.}$$

$$\sqrt{H} \propto DN$$

$$D \propto \sqrt{H}/N$$

$$\frac{P}{\left(\frac{\sqrt{H}}{N}\right)^2 \cdot H^{3/2}} = \text{const.}$$

$$\frac{N^2 P}{H^{5/2}} = \text{const.}$$

$$\frac{N\sqrt{P}}{H^{5/4}} = \text{constant} \quad * * *$$

$$\left( \frac{N\sqrt{P}}{H^{5/4}} \right)_1 = \left( \frac{N\sqrt{P}}{H^{5/4}} \right)_2$$

$$H_2 = 1\text{m} ; P_2 = 1\text{kW} ; N_2 = N_s$$

$$\frac{N\sqrt{P}}{H^{5/4}} = \frac{N_s \sqrt{1}}{(1)^{5/4}} = N_s$$

$$N_s = \frac{N\sqrt{P}}{H^{5/4}} \quad * \begin{array}{l} \rightarrow 10-35 \\ \rightarrow 60-300 \\ \rightarrow 300-900 \end{array}$$



$$N_s = \frac{N \sqrt{P}}{H^{5/4}}$$

$N$ : speed of runner (Rpm)  
 $H$ : Net Head (m)  
 $P$ : S.P in (KW)

Specific Head has Dimensions.

Non-Dimensional specific speed

$$N_s = \frac{N \sqrt{P/\rho}}{(gH)^{5/4}}$$

$H \rightarrow m$

$P \rightarrow W$

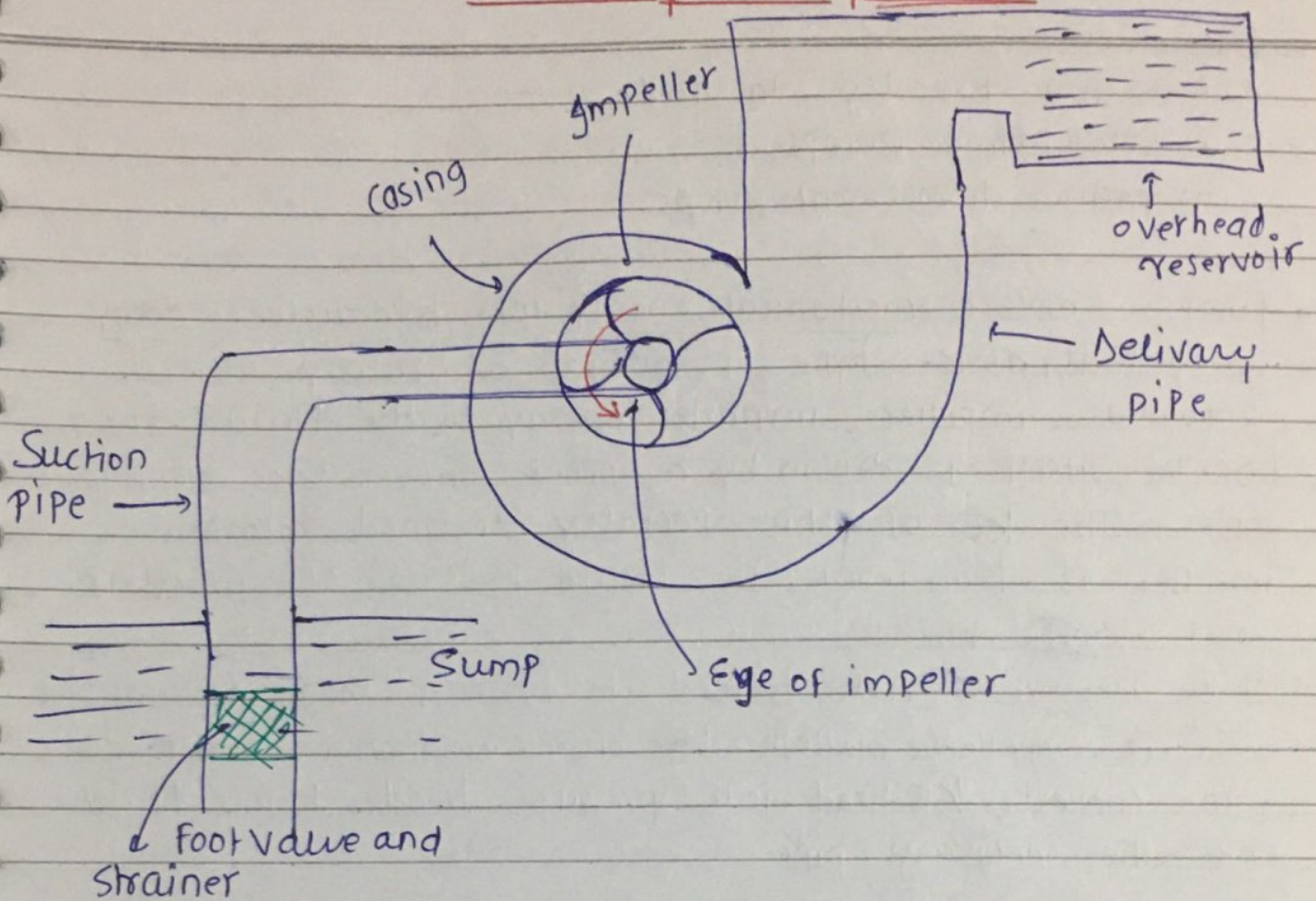
$N \rightarrow rpm$

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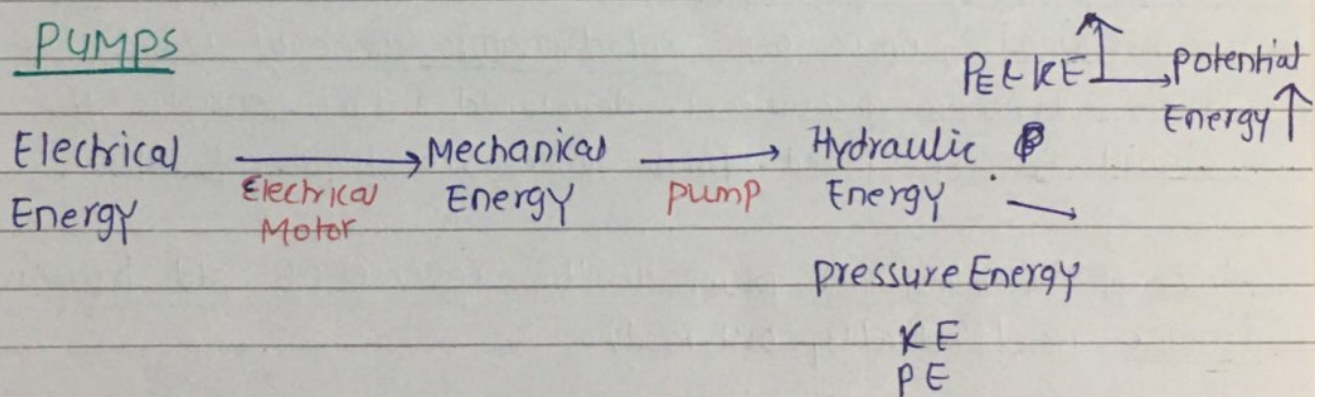
- Specific speed is same for all turbines of same type.
- specific Speed is related to shape & type of turbine but not it's size and therefore it is true characteristic of a turbine and hence classification of turbine on the basis of  $N_s$  is the most scientific criteria



# CENTRIFUGAL PUMPS



## Pumps



pumps are the devices which are used for transportation of a liquid from one location to another

Ex: pumping of water from a well or sump to an overhead reservoir

pumps are also used to circulate a liquid in a close system

Ex: circulation of cooling water and lubricating oil to various parts of a machine.



pumps are broadly classified as -

- i) Rotodynamic pumps
- ii) positive displacement pumps

- pumps Convert mechanical energy into hydraulic energy
- In rotodynamic type of pumps a rotary element known as impeller imparts energy to the fluid, the impeller itself is driven by a prime mover like electric motor. The KE of this specially designed rotating impeller is transferred to liquid in terms of pressure and kinetic energy.

It is however not desired to have a large amount of KE @ impeller outlet therefore arrangement is made to convert KE head into pressure head before fluid enters the delivery pipe.

Centrifugal pumps are rotodynamic type of pumps in which dynamic pressure is developed which enables the liquid to be lifted from lower to higher level

Action of Centrifugal pump is the reverse of old Francis turbine i.e (Radially outward)

Centrifugal pumps work on principle of forced Vortex motion in which when an external torque is applied, rise of pressure takes place in the rotating fluid. Which enables the lifting of liquid.



## Components

① Impeller It is the rotor with a series of curved vanes. It is mounted on a shaft coupled to electric motor. Electric motor imparts the required energy for rotating the impeller.

Impellers are of 3-types

① closed impeller

It is used for handling non-viscous liquids which are free from debris. It has better efficiency.

② semi-open impeller

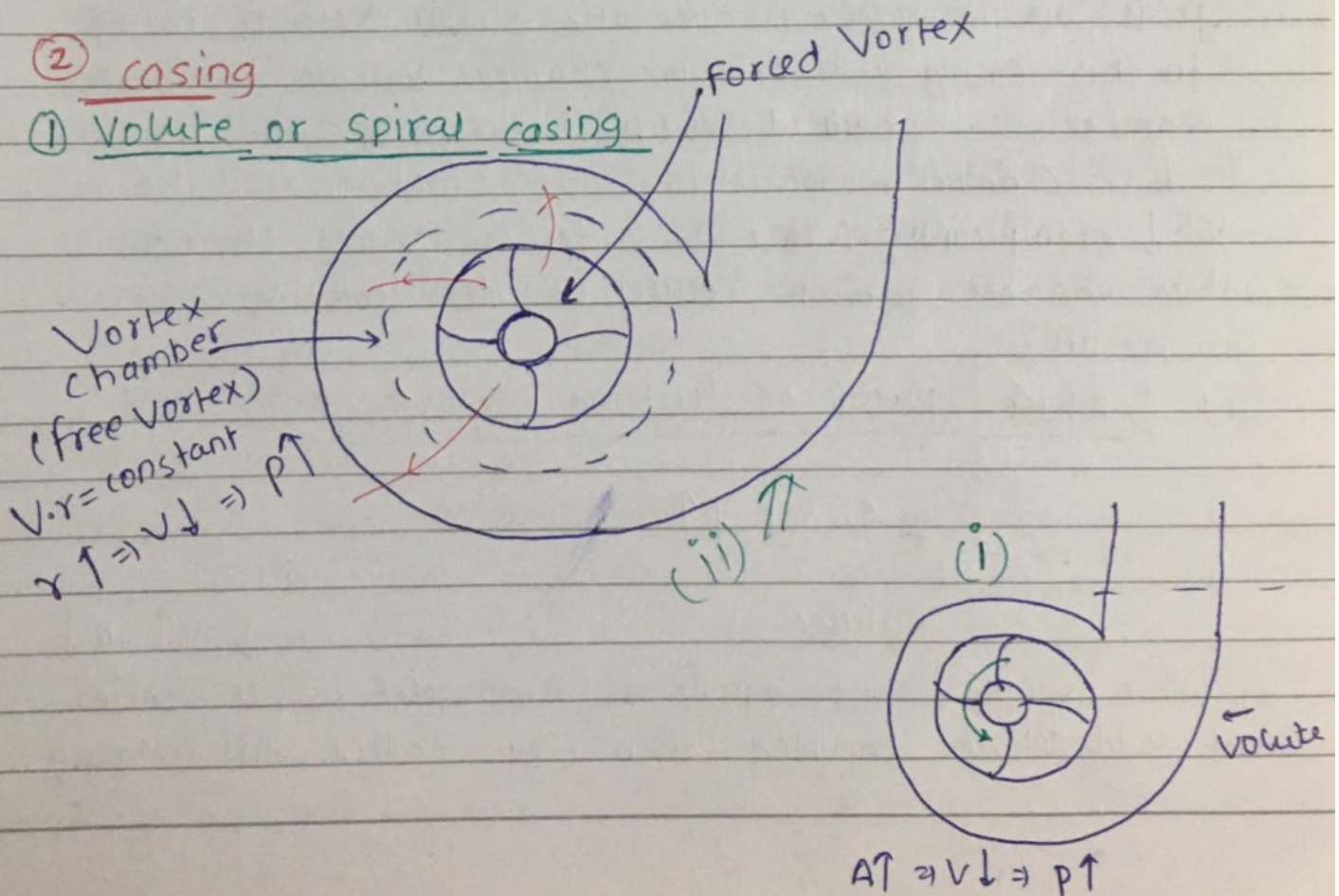
It is used for handling viscous fluids such as sewage water, paper pulp and liquids containing small amount of suspended particles.

③ open impeller

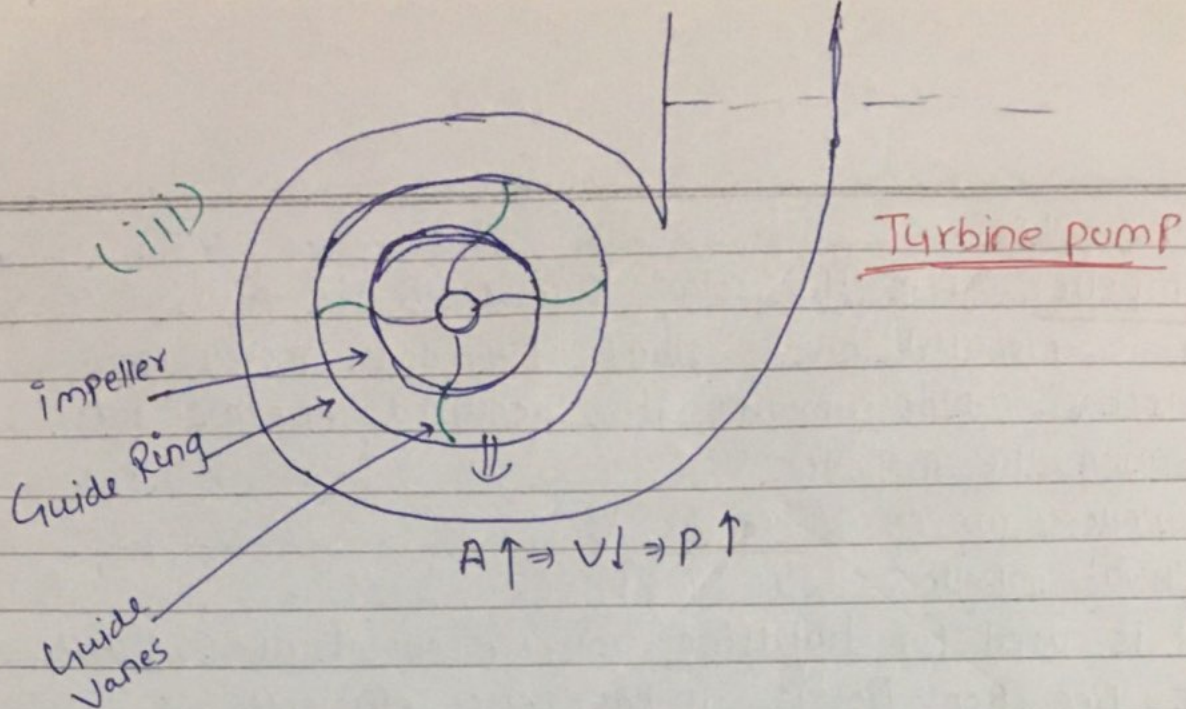
It is used to perform rough duties like handling abrasive liquids, mixture of water & sand.

② casing

① Volute or spiral casing





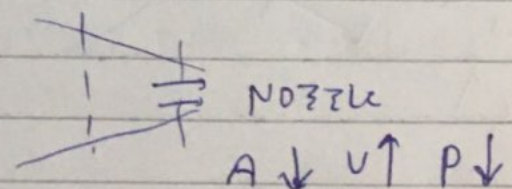
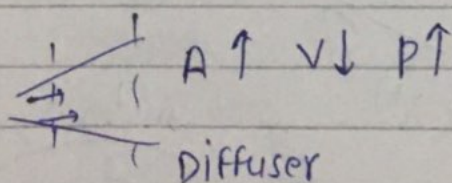


Shape of casing is such that Area of flow around periphery of impeller gradually increases towards the delivery side. This gradually increased area causes decrease in Velocity and hence increase in pressure.

## ii) Volute casing with Vortex chamber

It is an improvement over the simple Volute casing. In this casing, a circular chamber known as Vortex chamber is provided in b/w impeller and casing. In this chamber motion is free Vortex motion and hence  $KE \downarrow$  significantly while liquid is flowing through this chamber which results in corresponding rise in pressure.

## iii) Diffuser pump or turbine pump



In this type of casing impeller is surrounded by a series of guide blades mounted on a ring called diffuser ring.



Guide Vanes provide gradually increasing passage for the flow of fluid where Velocity  $\downarrow$  and pressure  $\uparrow$

Guide Vanes are so designed that liquid coming out of an impeller enters the guide passage smoothly. After the exit from guide passage water enters the casing which can be circular or spiral.

Based on number of entrance to impeller pumps can be single suction pump or Double suction pump. In single suction pump fluid is admitted from the suction pipe on one side of impeller and hence there is an axial thrust. However in double suction pump fluid enters from both sides of impeller and hence axial thrust is neutralised.

On the Basis of working head (centrifugal pumps can be classified as - ~~It~~ ~~Lift~~ Lift

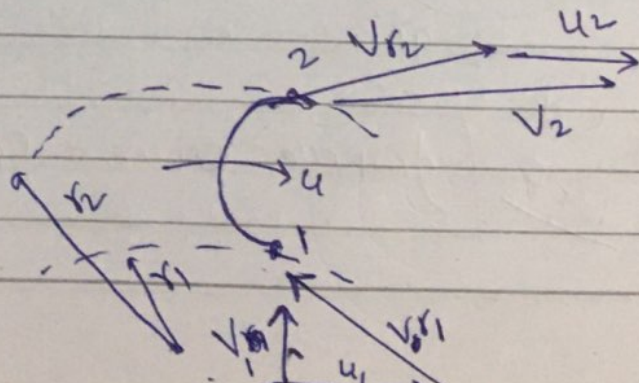
i) Low Lift pump • Head  $< 15\text{ m}$   
• impeller with Volute casing is used

ii) Medium Lift pump  $15 < H < 40\text{ m}$   
casing with diffuser ring is used

iii) High Lift pump  $H > 40\text{ m}$   
Multistage pumping is used

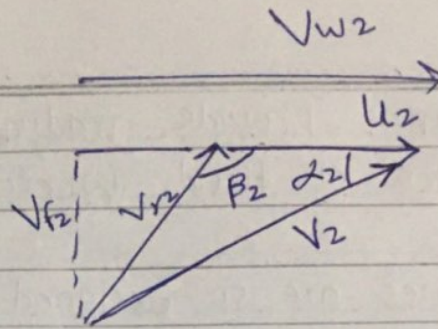
Impeller power

(Work done by impeller on water)

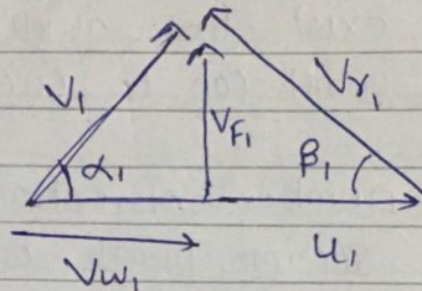




Exit velocity  $\Delta$ ngle



Inlet velocity  $\Delta$ ngle



Work done by impeller on water —

$$= \omega \times \text{Impeller on Water}$$

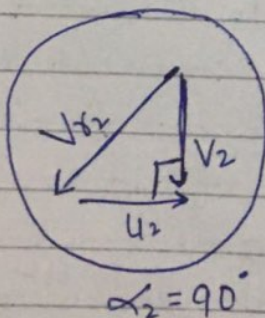
$$= \omega \times \text{Rate of change of angular momentum in clockwise dir}^n$$

$$= \omega \times \left( \text{Angular momentum flux out in } \vec{r} \text{ dir}^n - \text{Angular momentum flux in } \vec{r} \text{ dir}^n \right)$$

$$= \omega \times \left( \rho Q V_{w2} r_2 - \rho Q V_{w1} r_1 \right)$$

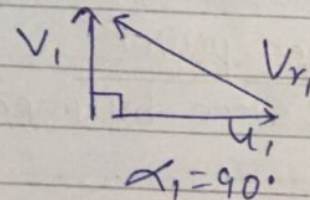
$$= \rho Q \left[ V_{w2} (\omega r_2) - V_{w1} (\omega r_1) \right]$$

$$\text{Impeller power} = \rho Q [V_{w2} U_2 - V_{w1} U_1]$$



$$\alpha_2 = 90^\circ$$

Discharge is radial @ entry



$$\alpha_1 = 90^\circ$$

$$V_1 = V_{f1} ; V_{w1} = 0$$

\*\*\*  
PRE

$$\text{Impeller power} = \rho Q V_{w2} U_2$$



$$\begin{aligned}\text{Euler's Head (H}_e\text{)} &= \frac{\text{Impeller power}}{\text{Wt. of fluid}} \\ &= \frac{\rho Q V w_2 u_2}{\rho Q g}\end{aligned}$$

$$\boxed{H_e = \frac{V w_2 u_2}{g}} *$$

practically it's observed that impeller efficiency is Max. When entry to impeller blade is radial i.e.  $\alpha_1 = 90^\circ$ .  $V w_1 = 0$  and  $V_1 = V_{r1}$

$$\begin{aligned}H_e &= \frac{V w_2 u_2 - V w_1 u_1}{g} \\ &= \underbrace{\frac{V_2^2 - V_1^2}{2g}}_{\Delta(\text{KE})} + \underbrace{\frac{u_2^2 - u_1^2}{2g} + \frac{V_{r1}^2 - V_{r2}^2}{2g}}_{\Delta(\text{press. Energy})}\end{aligned}$$

$\frac{V_2^2 - V_1^2}{2g} \rightarrow$  represents increase in KE of fluid

$\frac{u_2^2 - u_1^2}{2g} \rightarrow$  represents increase in pressure energy due to centrifugal action

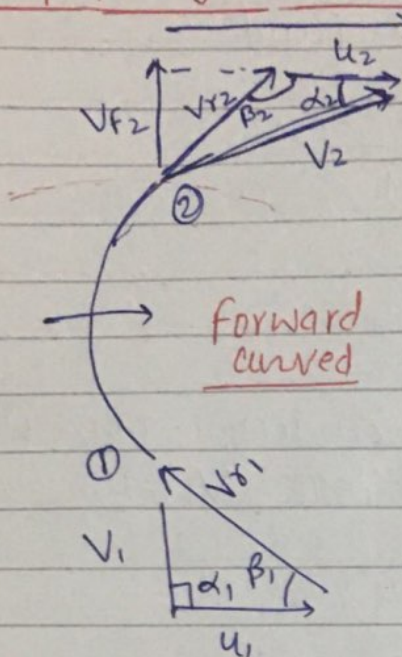
$\frac{V_{r1}^2 - V_{r2}^2}{2g} \rightarrow$  represents increase in pressure energy due to retardation of fluid while flowing through the blade passage.

\* Energy transfer takes place in impeller whereas energy transformation (KE to P.E) takes place in casing.



~~Not for PPT~~

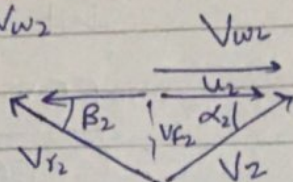
## Velocity Triangle at outlet in diff<sup>n</sup> cases



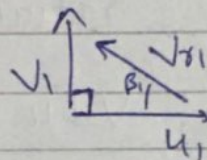
$$\beta_2 > 90^\circ$$

Slow Runner

$$Q \uparrow \rightarrow H \uparrow \rightarrow P \uparrow \uparrow$$



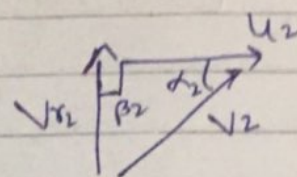
Backward  
Curved



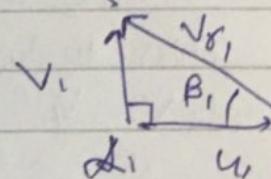
$$\beta_2 < 90^\circ$$

Fast Runner

$$Q \uparrow \rightarrow H \downarrow = P$$



Radial  
Blade



$$\beta_2 = 90^\circ$$

Medium speed  
Runner

IMP

- if outlet tip of blade curves in dir<sup>n</sup> of Blade motion then the Blade is said to be Forward curved blade ( $\beta > 90^\circ$ )  
Forward curved provide Very low efficiency.

- for forward curved blade with  $Q \uparrow$ , Head developed  $\uparrow$  but power req.  $\uparrow \uparrow$  exponentially and hence there is a chance of burn out of motor. therefore these are not used practically.

- if outlet tip of blade curves in dir<sup>n</sup> opposite to that of blade motion then blade is said to be Backward curved blade ( $\beta_2 < 90^\circ$ ), Fast Runner,

- $Q \uparrow$ , Head developed  $\downarrow$  and P req. remains nearly Constant.



These provides Max efficiency Hence most of the Centrifugal pumps are these only

- if outlet tip of blade makes an angle of  $90^\circ$  with the dir<sup>n</sup> of blade motion then it is called Radial blade
- medium speed runner

### Relation B/w Head developed and Discharge, power

Head imparted by impeller to water =  $V_{w2} u_2 / g$

Neglecting losses in pump

$$\text{Head developed} = \frac{V_{w2} u_2}{g}$$

from velocity triangle  $\tan \beta_2 = \frac{V_{f2}}{u_2 - V_{w2}}$

$$u_2 - V_{w2} = V_{f2} \cdot \cot \beta_2$$

$$V_{w2} = u_2 - V_{f2} \cot \beta_2$$

$$Q = \pi D B_1 \times V_{f1} = \pi D_2 B_2 \times V_{f2}$$

$$V_{f2} = \frac{Q}{\pi D_2 B_2}$$

$$V_{w2} = u_2 - \frac{Q}{\pi D_2 B_2} \cot \beta_2$$

$$H = \left( u_2 - \frac{Q}{\pi D_2 B_2} \cot \beta_2 \right) \cdot \frac{u_2}{g}$$

$$H = \frac{u_2^2}{g} - \frac{Q u_2}{\pi D_2 B_2 g} \cot \beta_2$$

$$H = K_1 - K_2 Q$$

$$K_1 = \frac{u_2^2}{g}$$

$$K_2 = \frac{u_2}{\pi D_2 B_2 g} \cot \beta_2$$

for forward curved blade,  $\beta_2 > 90^\circ$  ;  $\cot \beta_2 = -ve$

$K_2 = -ve$

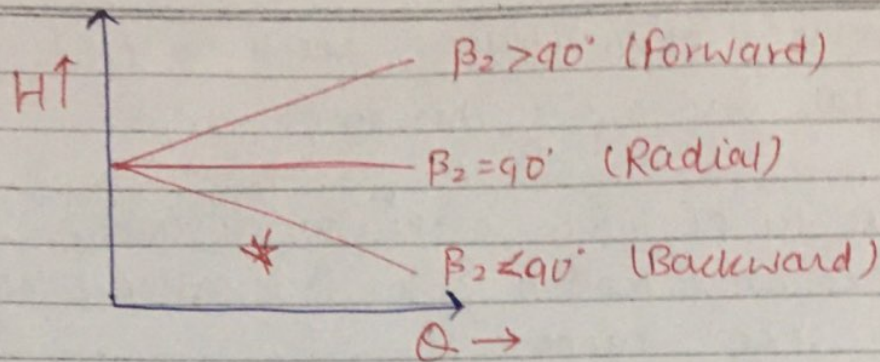
for Backward curved blade ;  $\beta_2 < 90^\circ$

$\cot \beta_2 = +ve$  ,  $K_2 = +ve$

for radial blade,  $\beta_2 = 90^\circ$

$\cot \beta_2 = 0$  ,  $K_2 = 0$





Neglecting losses,

$$\text{power input} = \rho g Q H$$

$$P \propto QH$$

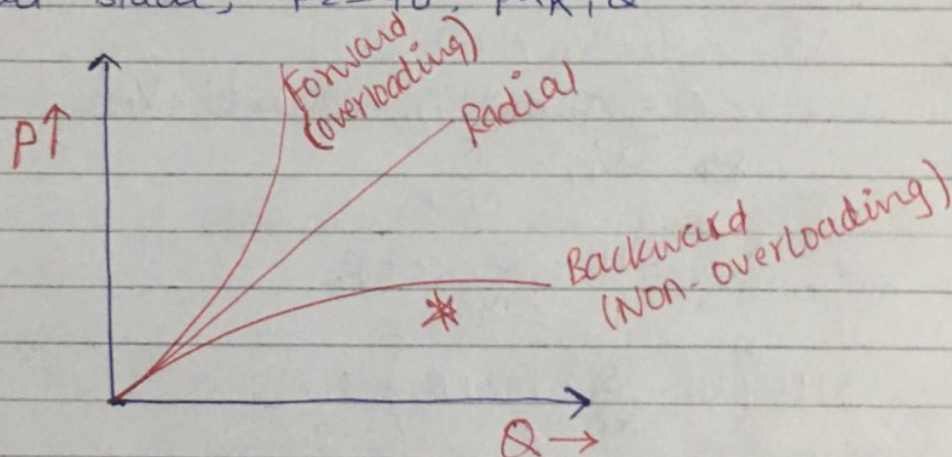
$$P \propto Q(K_1 - K_2 Q)$$

$$P \propto K_1 Q - K_2 Q^2$$

forward curved blade ;  $\beta_2 > 90^\circ \Rightarrow K_2 = -ve$

Backward curved blade ;  $\beta_2 < 90^\circ \Rightarrow K_2 = +ve$

Radial blade ;  $\beta_2 = 90^\circ$  ,  $P \propto K_1 Q$



if  $Q=0$

$$H = \frac{U_2^2}{g} \Rightarrow \text{Shut off Head}$$

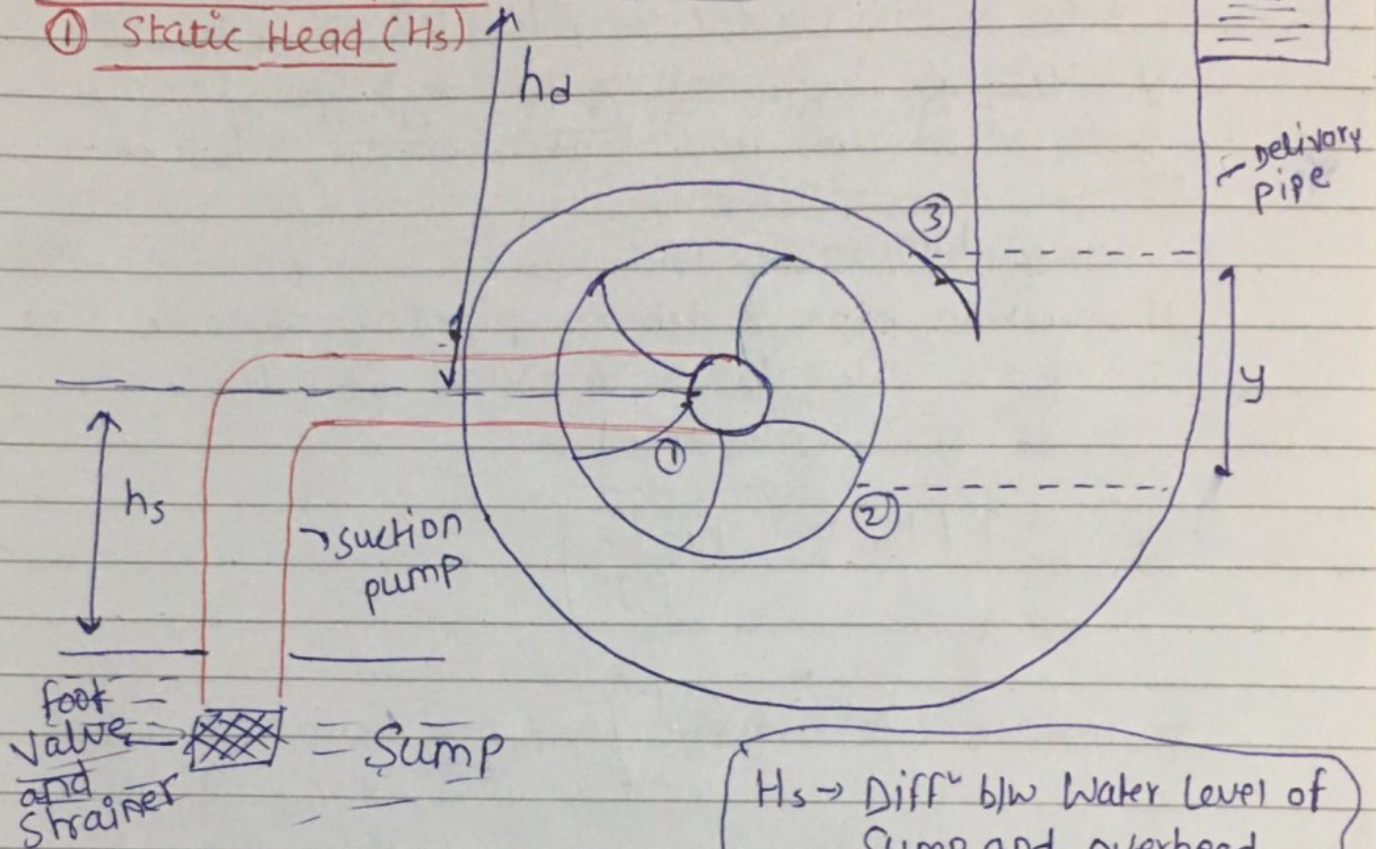
for a given angle  $\beta_2$ , theoretical Head  $H_e$  Varies Linearly with  $Q$ . When  $Q=0$  there is a finite +ve value of  $H_e$ . This indicates that if delivery valve of a running pump is completely closed shut and pump is kept running,



a +ve pressure head is produced by pump  
This is known as Shut off Head.

### Head of the pump

#### ① Static Head ( $H_s$ )



$h_s \rightarrow$  suction head  
 $h_d \rightarrow$  delivery head.

$$H_s = h_s + h_d$$

$H_s \rightarrow$  Diff<sup>n</sup> b/w Water Level of Sump and overhead reservoir where water is to be stored after pumping

#### ② Manometric head ( $H_m$ )

Demand  $H_m = h_s + h_d + h_{fs} + h_{fd} + \frac{V_d^2}{2g}$  — ①  
Net Head imparted by impeller to water

$$= \frac{V_{w2} u_2}{g} - h_{L,i} - h_{L,c}$$
 — ②

$$H_m = \frac{V_{w2} u_2}{g} - h_{L,i} - h_{L,c}$$



Applying energy eq<sup>n</sup> b/w points (1-1) and (2-2)

$$\frac{P_s}{\rho g} + \frac{V_s^2}{2g} + \frac{V_{w2} u_2}{g} = \frac{P_d}{\rho g} + \frac{V_d^2}{2g} + y + h_{l,i} + h_{l,c}$$

$$\underbrace{\frac{V_{w2} u_2}{g} - h_{l,i} - h_{l,c}}_{H_m} = \frac{P_d - P_s}{\rho g} + \frac{V_d^2 - V_s^2}{2g} + y$$

if suction pipe & delivery pipes are of same size.

$$d_s = d_d \Rightarrow V_s = V_d$$

if  $y$  is Neglected.

$$H_m = \frac{P_d - P_s}{\rho g}$$

$$= \left( \frac{\Delta P}{\rho g} \right)_{\text{pump}}$$

$$= \left( \frac{\Delta P}{\rho g} \right)_{\text{impeller}} + \left( \frac{\Delta P}{\rho g} \right)_{\text{casing}}$$

$$\begin{matrix} P_3 > P_2 \\ V_3 < V_2 \end{matrix}$$

$$\begin{matrix} P_2 > P_1 \\ V_2 > V_1 \end{matrix}$$

$$V_{r2} < V_{r1}$$

①  $H_m \rightarrow$  it is the total head that must be produced by pump to satisfy external req.

The external req. is to lift the water to the req level ( $H_s$ )

Meet the energy req. of losses in suction and delivery pipe and to deliver the water @ a velocity  $V_d$  ~~req.~~

hence we get the eq<sup>n</sup> ①



This external req. has to be met by pump hence net head imparted by pump must cater for manometric head. That is eq<sup>n</sup> ②

point ① is in Suction Pipe Very close to pump

point ② is @ exit of impeller.

point ③ is in delivery pipe Very close to the pump

pressure rise across impeller

Applying Energy eq<sup>n</sup> b/w inlet and exit of impeller

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + Z_1 + \frac{V_{w1} u_1}{g} = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + Z_2 + h_{L,i}$$

$$\frac{P_2 - P_1}{\rho g} = (\Delta P)_i = \frac{V_1^2}{2g} - \frac{V_2^2}{2g} - h_{L,i} + \frac{V_{w1} u_1}{g}$$

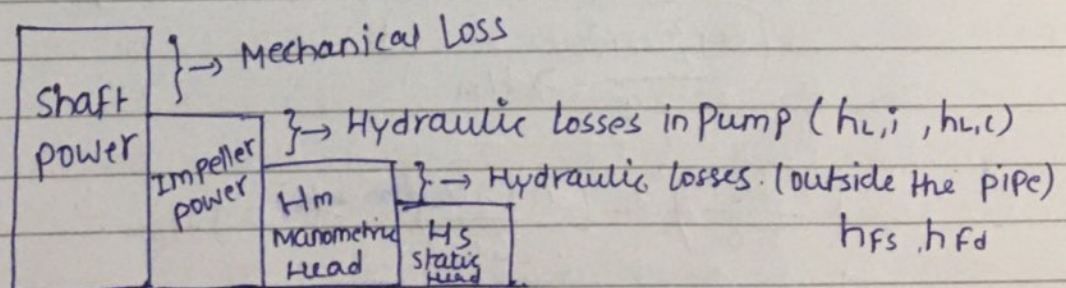
pressure rise in casing

Applying Energy eq<sup>n</sup> b/w entry and exit of casing

$$\frac{P_2}{\rho g} + \frac{V_2^2}{2g} = \frac{P_d}{\rho g} + \frac{V_d^2}{2g} + y + h_{L,c}$$

$$\left( \frac{P_d - P_2}{\rho g} \right) = (\Delta P)_c = \frac{V_2^2 - V_d^2}{2g} - y - h_{L,c}$$

Losses and Efficiencies





1) Mechanical Efficiency  $\eta_m = \frac{\text{impeller power}}{\text{shaft power}}$

2) Manometric Efficiency  $\eta_{\text{mano}} = \frac{\text{Head developed (Hm)}}{\text{Head imparted by impeller to Water (He)}}$

$$= \frac{H_m}{\frac{V_{w2} u_2}{g}} \Rightarrow \boxed{\eta_{\text{mano}} = \frac{g H_m}{V_{w2} u_2}}$$

3) Volumetric efficiency  $\eta_{\text{vol}} = \frac{\text{Quantity of water delivery by pump}}{\text{Quantity of water passing through impeller}}$

4) Overall efficiency  $\eta_o = \frac{\rho g Q H_m}{\text{shaft power}}$

### Minimum starting speed of pump to deliver liquid:-

When the pump is started there will be no flow of water until the head imparted by impeller is large enough to overcome the manometric head

$$H_e \geq H_m$$

$$\left( \frac{V_2^2 - V_1^2}{2g} \right) + \left( \frac{u_2^2 - u_1^2}{2g} \right) + \left( \frac{V_{r1}^2 - V_{r2}^2}{2g} \right) \geq H_m$$

When the flow has not started;  $V_1 = 0, V_2 = 0, V_{r1} = 0, V_{r2} = 0$

$$\frac{u_2^2 - u_1^2}{2g} \geq H_m$$

$$\left( \frac{\pi D_2 N}{60} \right)^2 - \left( \frac{\pi D_1 N}{60} \right)^2 \geq 2g H_m$$

$$\boxed{N \geq \sqrt{\frac{2g H_m}{\left( \frac{\pi}{60} \right)^2 (D_2^2 - D_1^2)}}$$



## Diameter of Impeller and pipes

inlet      outlet  
↓          ↓  
inner      outer

(1) Outer dia  
of impeller ( $D_2$ )

$$U_2 = \frac{\pi D_2 N}{60} = K_u \sqrt{2gH_m}$$

(2) Inner Dia  
of impeller ( $D_1$ )

$$\Rightarrow \boxed{D_1 = \frac{D_2}{2}}$$

(3) Dia of Suction pipe ( $d_s$ )

$$Q = \frac{\pi}{4} d_s^2 \cdot V_s \Rightarrow d_s = \sqrt{\frac{4Q}{\pi V_s}}$$

4) Diameter of delivery pipe ( $d_d$ )

$$Q = \frac{\pi}{4} d_d^2 \cdot V_d \Rightarrow d_d = \sqrt{\frac{4Q}{\pi V_d}}$$

5) Least dia of impeller ( $D_{2,min}$ )

$$\frac{U_2^2 - U_1^2}{2g} \geq H_m \Rightarrow \left( \frac{\pi D_2 N}{60} \right)^2 - \left( \frac{\pi D_1 N}{60} \right)^2 \geq 2gH_m$$

$$\left( \frac{\pi N}{60} \right)^2 D_2^2 \left( 1 - \frac{1}{4} \right) \geq H_m$$

$D_1 = \frac{D_2}{2}$

Que: A centrifugal pump lifts 100 l/sec of a liquid against a net head of 150 kPa. Break power req. is 18 kW when the liquid is water with a specific wt. of  $9.8 \text{ kN/m}^3$ . What would be the break power if the liquid is a solvent having a relative density of 0.8 instead of water with all other factors remaining the same as before.

Ans: 18 kW



Cavitation if  $P_{abs} < P_{vap}$  cavitation occurs. (may)  
cavitation can be noted experimentally by sudden drop  
in efficiency and head.

In centrifugal pumps pressure is Suction pressure @ the entry to  
the pump and hence chances of cavitation are Max. @ the entry  
to the pump.

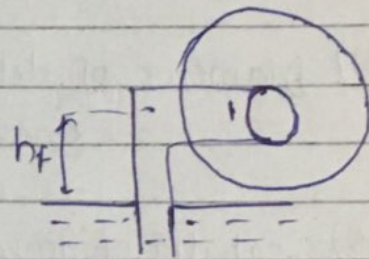
for no cavitation

$$\left(\frac{P_1}{\rho g}\right)_{abs.} > \left(\frac{P_{vap}}{\rho g}\right)$$

Applying Energy EQ<sup>n</sup> b/w inlet of pump and Surface of Sump

$$\frac{P_{atm}}{\rho g} = \frac{P_1}{\rho g} + \frac{V_s^2}{2g} + h_s + h_{fs}$$

$$\frac{P_1}{\rho g} = \frac{P_{atm}}{\rho g} - \frac{V_s^2}{2g} - h_s - h_{fs}$$



for no cavitation  $\frac{P_1}{\rho g} > \frac{P_{vap}}{\rho g}$

$$\boxed{\frac{P_{atm}}{\rho g} - \frac{V_s^2}{2g} - h_s - h_{fs} > \frac{P_{vap}}{\rho g}}$$

IMP.

chances of cavitation can be reduced by

- i)  $P_{atm} \uparrow$
- ii)  $V_s \downarrow \Rightarrow d_s \uparrow$
- iii)  $h_s \downarrow$
- iv)  $h_{fs} \downarrow$   $\rightarrow$  should be
- v)  $P_{vap} \downarrow$  ( $T \downarrow$ )

Max. Suction lift ( $h_{s,max}$ )

$$h_s < \frac{P_{atm} - P_v}{\rho g} - \frac{V_s^2}{2g} - h_{fs}$$

$$(h_s)_{max} = \frac{P_{atm} - P_v}{\rho g} - \frac{V_s^2}{2g} - h_{fs}$$



for Water  $\frac{P_v}{P_g} = 0.5m$

$\frac{P_{atm}}{P_g} = 10.3$

6-8 m

Net positive suction head (NPSH)

$$(NPSH)_{avail} = \left( \frac{P_s}{P_g} + \frac{V_s^2}{2g} \right) - \frac{P_v}{P_g}$$

$$= \frac{P_{atm}}{P_g} - h_s - h_{fs} - \frac{P_v}{P_g}$$

\* 
$$(NPSH)_{avail} = \frac{P_{atm} - P_v}{P_g} - h_s - h_{fs}$$

for No cavitation;

$$(NPSH)_{avail} > (NPSH)_{req.}$$

(NPSH)<sub>available</sub> is the energy available, in forms of pressure & KE @ the inlet to the pump above Vapor pressure of water. it is a measure of margin against Liquid Vaporization, and hence Cavitation Susceptibility

(NPSH)<sub>req.</sub> is decided by Manufacturer.

Thoma's Cavitation Factor ( $\sigma$ )

$$\sigma = \frac{NPSH}{H_m}$$

for no cavitation

$$\sigma > \sigma_c$$

(critical Cavitation factor =  $f(N_s)$ )

$N_s \uparrow \Rightarrow \sigma_c \uparrow \Rightarrow$  chances of Cavitation.

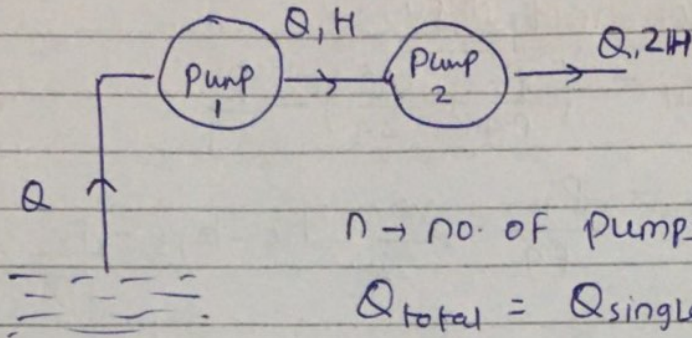


## multistage pumping

i) pumps in series

$Q$  remains same

$H \uparrow$

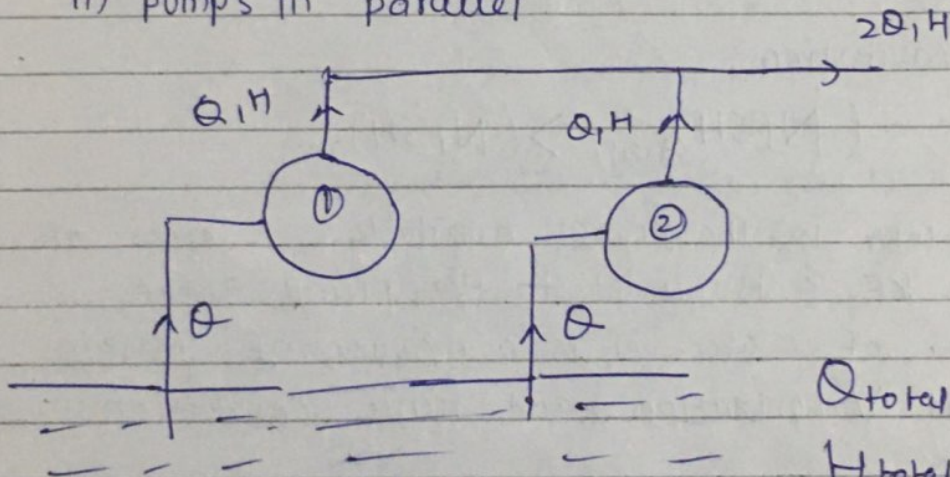


$n \rightarrow$  no. of pumps

$$Q_{\text{total}} = Q_{\text{single}}$$

$$H_{\text{total}} = n \times H_{\text{single}}$$

ii) pumps in parallel



$$Q_{\text{total}} = n \times Q_{\text{single}}$$

$$H_{\text{total}} = H_{\text{single}}$$

## Specific Speed ( $N_s$ )

speed of a geometrically similar pump so reduced in size that it would deliver  $1 \text{ m}^3$  of liquid per second against a head of  $1 \text{ m}$ .

Water condition of max. efficiency—

$$\frac{QH}{N^2 D^2} = \text{const.}, \quad \frac{Q}{ND^3} = \text{const.}$$

$$\frac{P}{N^3 D^5} = \text{const.}$$



|                                                                                                                                                                                 |                                                                                                                                                                                                                                                                                                     |
|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| $\frac{Q}{D^3 N} = \text{const.}$ $\frac{\sqrt{H}}{DN} = \text{const}$ $\frac{Q}{\left(\frac{\sqrt{H}}{N}\right)^3 N} = \text{const.}$ $\frac{N^2 Q}{(H)^{3/2}} = \text{const}$ | $\frac{N\sqrt{Q}}{(H)^{3/4}} = \text{const.}$ $\left( \frac{N\sqrt{Q}}{(H)^{3/4}} \right) = \left( \frac{N\sqrt{Q}}{(H)^{3/4}} \right)$ <p>if <math>Q_2 = 1 \text{ m}^3/\text{s}</math> ; <math>H_2 = 1 \text{ m}</math></p> $N_2 = N_s$ $\frac{N\sqrt{Q}}{H^{3/4}} = \frac{N_s \sqrt{1}}{H^{3/4}}$ |
|---------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|

$$N_s = \frac{N\sqrt{Q}}{(H_m)^{3/4}}$$

$N$ :- speed of pump in rpm

$Q$ :- Discharge in  $\text{m}^3/\text{s}$  (lit/s)

$H_m$ :- Manometric Head in (m)

Non-Dimensional Specific Speed (Shape Number)

$$N_s = \frac{N\sqrt{Q}}{(gH_m)^{3/4}}$$

$N \rightarrow \text{rad/s}$

$Q \rightarrow \text{m}^3/\text{s}$

$H_m \rightarrow \text{m}$

While calculating  $N_s$  of a double Suction pump,  $Q$  is taken as half of total discharge

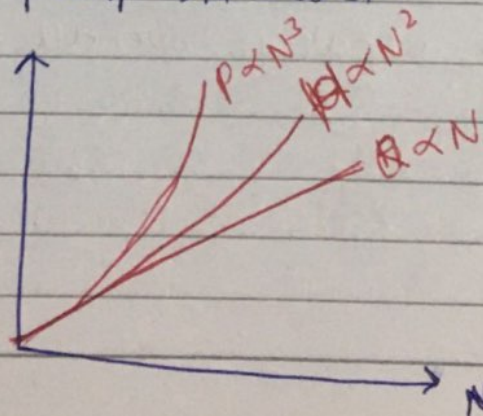
Main characteristic curve

for same pump  $Dia = \text{const.}$

$$H \propto N^2$$

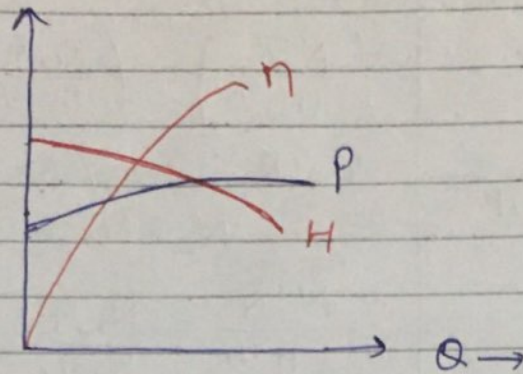
$$Q \propto N$$

$$P \propto N^3$$





### operating characteristic curve



### priming

Before starting a pump its impeller and suction pipe have to be filled with water to remove any air or gas from the waterways of pump. This is known as priming. If pump casing is not filled with water @ start of pump, impeller rotates in air filled casing this can't create the necessary differential head of water b/w sump and inlet as density of air is much less than that of water. Consequently the pump doesn't do its work of pumping of water.

A 1-way ~~gate~~ <sup>valve</sup> called foot valve is provided @ entrance of suction pipe to prevent the water from draining out ~~the~~ to sump when a running pump is closed.

Que: 2 centrifugal pumps worked in parallel with a common delivery head of 18m. First pump delivers 90 l/s @ an efficiency of 60%, 2<sup>nd</sup> pump delivers 60 l/s @ an efficiency of 50%. What is overall efficiency of system?

- A) 62.5%
- B) 55.5%
- C) 58%
- D) 52%



$$H_1 = H_2 = H_{\text{Total}} = 18 \text{ m}$$

$$Q_{\text{total}} = Q_1 + Q_2$$

$$= 90 + 60$$

$$= 150 \text{ LPS.}$$

$$\text{Power input to the 1st pump} = \frac{\rho g Q_1 H}{\eta_1}$$

$$\text{Power input to 2nd pump} = \frac{\rho g Q_2 H}{\eta_2}$$

$$\eta_o = \frac{\rho g Q_{\text{total}} H_{\text{total}}}{P_{i/p_1} + P_{i/p_2}} = \frac{\rho g Q_{\text{total}} H}{\frac{\rho g Q_1 H}{\eta_1} + \frac{\rho g Q_2 H}{\eta_2}}$$

$$= \frac{150}{\frac{90}{60} + \frac{60}{50}}$$

$$= 55.5\%$$

Que. Determine the no. of impellers req<sup>r</sup> for a multistage pump to lift 4000 lit/minute against a total head of 80 m. @ a speed of 750 rpm. Given that specific speed for each impeller should be b/w 720 - 780 units  
A) 6    B) 5    C) 4    D) 3

$$N_s = \frac{N \sqrt{Q}}{H^{3/4}}$$

$$750 = \frac{750 \sqrt{4000/60}}{H^{3/4}} \Rightarrow H = 16.4 \text{ m}$$

pumps are connected in series \*

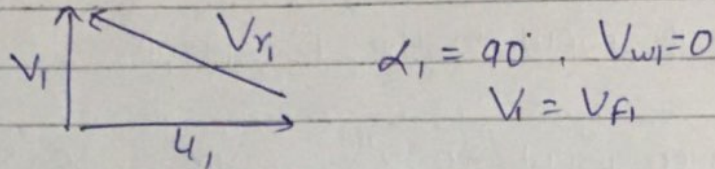
$$\text{No. of impellers req<sup>r</sup>} = \frac{80}{16.4} \approx 5$$



## Assumption

Unless o/w mentioned, it is assumed that

i) Absolute Velocity is Radial @ the inlet



ii) Velocity of Flow is constant.

Q. Water enters Radially through a Centrifugal pump  
whose impeller has dia = 30 cm and width = 15 cm  
the corresponding Dimension @ outer periferi are 60 cm  
and 7.5 cm, The blades are Curved backward @  $30^\circ$   
to tangent @ exit, Discharge is 0.235 Cumec.  
if the rotational Speed of impeller is 1200 Tpm.  
and the pump operates under the manometric head of  
115 m. Calculate ① Theoretical head developed and the  
Manometric efficiency ② The pressure rise  
across impeller, assuming losses = 10% of Velocity  
head @ exit. ③ pressure rise and loss of head in  
Volute casing ④ Vane angle @ inlet. ⑤ assuming  
Overall efficiency = 65%, Calculate the power req.  
to drive the pump and the corresponding mechanical  
efficiency.

i) Theoretical head developed ( $H_e$ )

$$H_e = \frac{V_{w2} u_2}{g}$$

$$u_2 = \frac{\pi D_2 N}{60} = \frac{3.14 \times 0.6 \times 1200}{60} = 37.7 \text{ m/s}$$



$$\underline{V_{w2}} \Rightarrow \tan \beta_2 = \frac{V_{f2}}{U_2 - V_{w2}}$$

$$V_{f2} = \frac{Q}{\pi D_2 B_2} = \frac{0.225}{3.14 \times 0.6 \times 0.075} = 1.59 \text{ m}$$

$$\tan 30^\circ = \frac{1.59}{37.7 - V_{w2}}$$

$$V_{w2} = 34.94 \text{ m/s}$$

$$H_e = \frac{V_{w2} U_2}{g} = \frac{34.94 \times 37.7}{9.81} = 134.27 \text{ m}$$

$$\eta_{\text{mano.}} = \frac{H_m}{H_e} = \frac{115}{134.27} = 85.64\%$$

ii) pressure rise across impeller

Apply Energy Eq<sup>n</sup> b/w ① & ②

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + Z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + Z_2 + H_{f,i} + H_e$$

$$\left( \frac{P_2 - P_1}{\rho g} \right) \rightarrow \text{pressure rise across impeller.}$$

$$\left( \frac{P_2 - P_1}{\rho g} \right) = \frac{V_1^2}{2g} - \frac{V_2^2}{2g} + H_e - 0.1 \frac{V_2^2}{2g}$$

$$V_1 = V_{f1} = V_{f2} = 1.592 \text{ m/s}$$

$$V_2 = \sqrt{V_{w2}^2 + V_{f2}^2} = \sqrt{34.94^2 + 1.592^2} = 34.978 \text{ m/s}$$

$$(\Delta P)_i = 65.814 \text{ m}$$

$$\text{iii)} \quad H_m = \frac{(\Delta P)_i}{\rho g} + \frac{(\Delta P)_c}{\rho g}$$

$$115 \text{ m} = \frac{65.814}{9.81} + \frac{(\Delta P)_c}{9.81} \Rightarrow (\Delta P)_c = 49.19 \text{ m}$$



$$H_m = \frac{V_{w2} u_2}{g} - h_{L,i} - h_{L,c}$$

$$115 = 134.27 - 6.23 - h_{L,c}$$

$$h_{L,c} = 13.05 \text{ m}$$

iv)  $\beta_1 = ?$

$$\tan \beta_1 = \frac{V_{F1}}{u_1} = \frac{1.592}{u_2/2}$$

$$\beta_1 = 4.83^\circ$$

v) power req<sup>r</sup> to drive the pump =  $\frac{\rho g Q H_m}{\eta_o}$   
 $= 390.15 \text{ kW}$

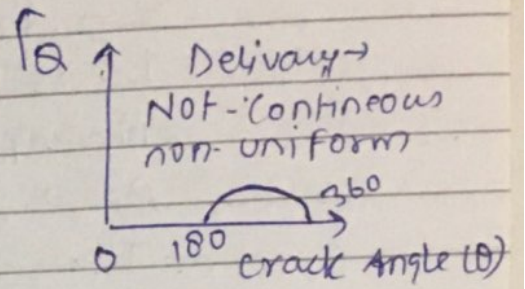
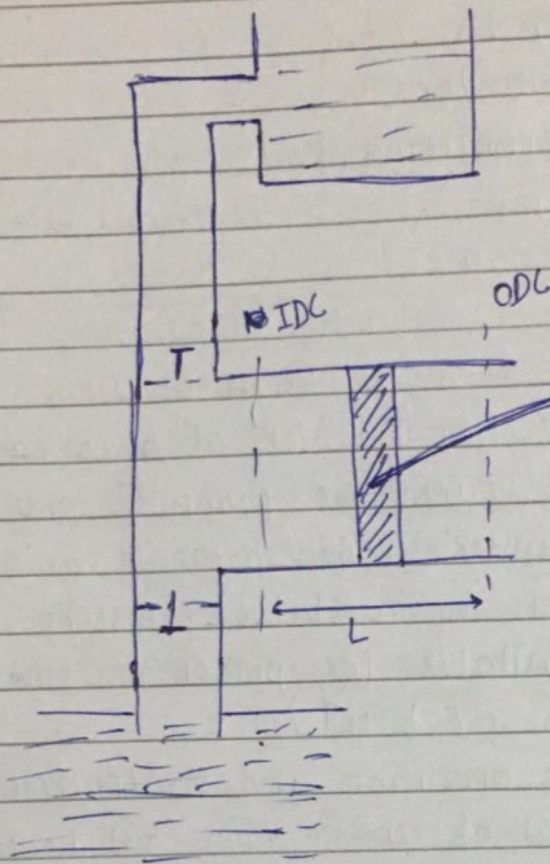
$$\eta_o = 85.64 \times \eta_{\text{mech}}$$

$$\eta_{\text{mech}} = 75.89 \%$$

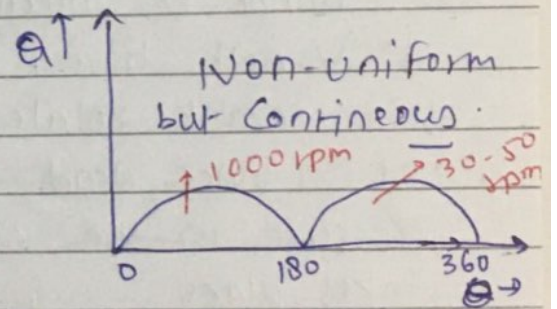
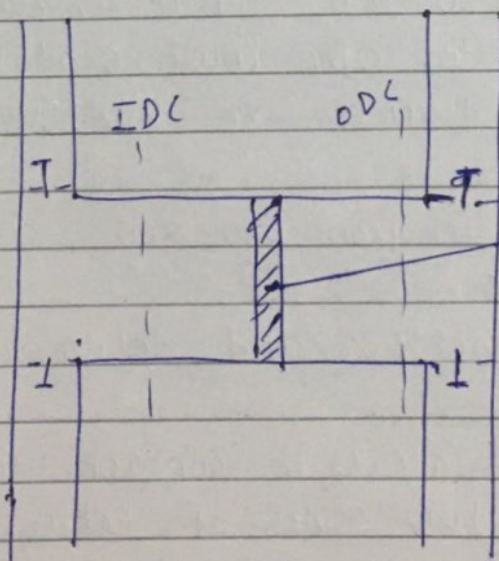
Ans



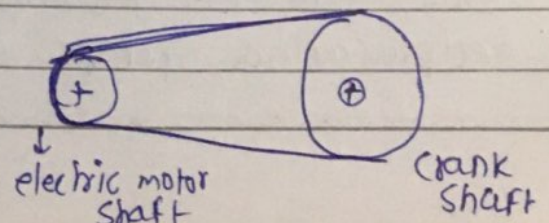
# Reciprocating pump



Single Acting reciprocating pump



Double Acting Reciprocating pump





$\theta$   $\rightarrow$  Angle turned by crank

$r$   $\rightarrow$  crank radius

$L$   $\rightarrow$  Stroke Length

$A$   $\rightarrow$  Area of cylinder

$A_p$   $\rightarrow$  Area of connecting rod

$a$   $\rightarrow$  Area of pipe

$l$   $\rightarrow$  Length of pipe

In reciprocating pumps or +ve displacement pumps energy transfer is accomplished by changing Vol<sup>m</sup> of a certain mass of fluid contained within the pump by physically displacing one of its boundaries. This increase in energy is manifested in the form of rise in static pressure.

Reciprocating pumps are suitable for pumping smaller quantity @ greater heights. ( $\theta \downarrow H \uparrow$ )

It is used in oil drilling operation and in concrete pumps. It works @ a low speed of 30-50 rpm and hence it cannot be directly coupled to the electric motor.

A V-belt drive is used to reduce the speed.

When crank rotates from  $0-180^\circ$ , piston moves from IDC (inner dead center) to ODC (outer dead center). Pressure in the cylinder becomes -ve and suction takes place.

As the crank rotates ~~piston~~ from  $180-360^\circ$ , piston moves from ODC  $\rightarrow$  IDC.

pressure in cylinder becomes +ve and delivery takes place.

Depending on whether water is in contact with one side of piston or two sides of piston, reciprocating pumps can be classified as ~~two~~



i) single acting

ii) Double Acting.

\* In single acting pump Discharge is non-continuous as well as non-uniform. Whereas in Double acting pump Discharge is continuous ~~and~~ But non-uniform.

power input in Reciprocating pump

for single Acting pump

$$Q_{th} = \frac{ALN}{60}$$

$$P_{ip} = \rho g Q_{th} (h_s + h_d)$$

$$* P_{ip} = \rho g \left( \frac{ALN}{60} \right) (h_s + h_d) \rightarrow \text{Theoretical}$$

for Double Acting pump

$$Q = \frac{ALN}{60} + \frac{(A - A_p)LN}{60}$$

if  $A_p$  is neglected

$$* P_{ip} = \rho g \left( \frac{2ALN}{60} \right) (h_s + h_d)$$

This is the theoretical power input because here losses have not been considered

-ve slip

Slip, percentage slip, coefficient of Discharge ( $C_d$ )

$$\text{slip} = Q_{th} - Q_{actual}$$

$$\% \text{ slip} = \frac{Q_{th} - Q_{act}}{Q_{th}} \times 100$$

$$C_d = \frac{Q_{actual}}{Q_{th}}$$

if  $Q_{actual} > Q_{th}$   
 $C_d > 1$

Slip  $\rightarrow$  -ve.



-ve slip occurs, When Suction pipe is long, delivery pipe is short and speed of rotation is high.

In this case inertia force in Suction pipe will be large in comparison to force outside delivery valve due to which, delivery Valve opens, before Suction Stroke is completed. Thus Some of liquid is pushed directly from Suction pipe to delivery pipe and hence pump will discharge more quantity than the swept Vol<sup>m</sup> of pump

Effect of acceleration of piston on Velocity and pressure in Suction and delivery pipe -

$x$  = Displacement of piston

$$= r - r \cos \theta$$

$$= r - r \cos \omega t$$

Velocity of piston;  $V_p = \frac{dx}{dt}$

$$\frac{dx}{dt} = 0 - r(-\omega \sin \omega t)$$

$$V_p = r \omega \sin \omega t$$

~~acceleration of piston~~  $A_p$

Velocity of water in cylinder =  $r \omega \sin \omega t$

Velocity of Water in pipe ( $V$ ) =

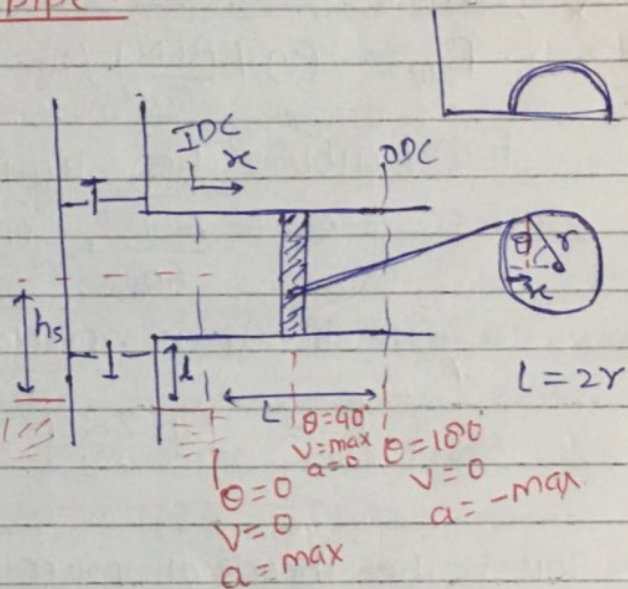
$$A \times r \omega \sin \omega t = a \times V$$

$$V = \frac{A}{a} r \omega \sin \omega t$$

Acceleration of water in pipe

$$\frac{d}{dt} \left( \frac{A}{a} r \omega \sin \omega t \right)$$

$$= \frac{A}{a} r \omega^2 \cos \omega t$$





$$\text{mass of water in pipe} \times \text{Acceleration of water in pipe} \\ = \rho a l \times \frac{A}{a} r \omega^2 \cos \omega t = \rho \times a$$

$$\rho = \frac{A}{a} \rho r \omega^2 \cos \omega t$$

$$H_a = \frac{P}{\rho g} = \frac{A}{a} \frac{l}{g} r \omega^2 \cos \omega t$$

$$H_a = \frac{A}{a} \frac{l}{g} r \omega^2 \cos \omega t \quad * \quad \text{Acceleration head.}$$

Acceleration head during suction stroke

$$H_{as} = \frac{A}{a_s} \frac{l_s}{g} r \omega^2 \cos \omega t$$

Acceleration head during delivery stroke

$$H_{ad} = \frac{A}{a_d} \frac{l_d}{g} r \omega^2 \cos \omega t$$

| Suction stroke               | Velocity of water in pipe | Acceleration of water in pipe |
|------------------------------|---------------------------|-------------------------------|
| Beginning ( $\theta=0$ )     | 0                         | MAX (Acc)                     |
| Middle ( $\theta=90^\circ$ ) | Max                       | 0                             |
| End ( $\theta=180^\circ$ )   | 0                         | Max (retardation)             |

\*  $H_{as}$  is max. @ beginning of suction stroke ~~retardation~~

Effect of Variation in Velocity in pipe on friction in pipe

$$\text{Frictional losses in pipe} = \frac{f l v^2}{2 g d}$$

$f \rightarrow$  friction factor

$l \rightarrow$  length of pipe

$d \rightarrow$  Dia of pipe

$v \rightarrow$  velocity of water



### Suction stroke

$$h_{fs} = \frac{f l_s v_s^2}{2 g d_s} = \frac{f l}{2 g d_s} \left( \frac{A}{a_s} r \omega \sin \omega t \right)^2$$

$\begin{cases} \rightarrow 0 & \text{beginning} \\ \rightarrow \text{max.} & \text{middle} \\ \rightarrow 0 & \text{End} \end{cases}$

$$(h_{fs})_{\max} = \frac{f l}{2 g d_s} \left( \frac{A}{a_s} r \omega \right)^2$$

### Delivery stroke

$$h_{fd} = \frac{f l_d}{2 g d_d} \left( \frac{A}{a_d} r \omega \sin \omega t \right)^2$$

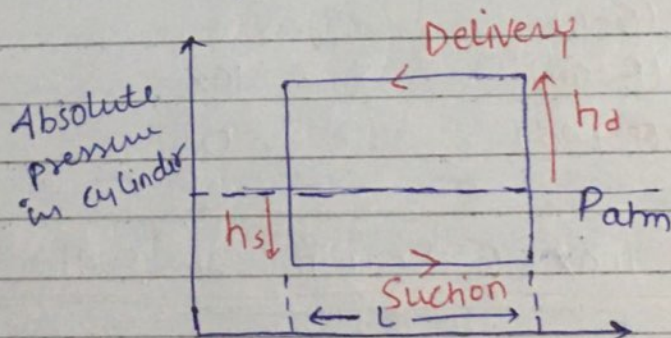
$\begin{cases} \rightarrow 0 & \text{Beginning} \\ \rightarrow \text{max.} & \text{middle} \\ \rightarrow 0 & \text{End.} \end{cases}$

$$(h_{fd})_{\max} = \frac{f l_d}{2 g d_d} \left( \frac{A}{a_d} r \omega \right)^2$$

**NOTE** Effect of friction head is max. @ middle of Suction Stroke and Middle of delivery Stroke

### Indicator Diagram

It is the graph b/w absolute pressure in cylinder and Stroke length



Power input to reciprocating pump

$$P = \rho g \left( \frac{A L N}{60} \right) (h_s + h_d)$$

$$= \rho g \frac{A N}{60} L (h_s + h_d)$$

$$= K L (h_s + h_d)$$

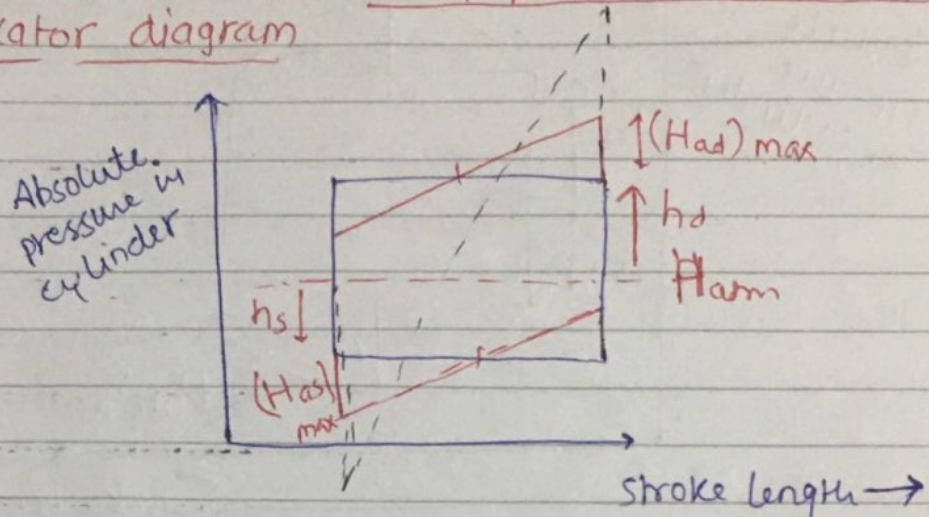
$$= K \times \text{Area of indicator diagram}$$

stroke length  $L \rightarrow$   
Ideal indicator diagram  $\Theta \rightarrow$



\* Area of indicator Diagram to some scale represents power input in a reciprocating pump

Effect of acceleration in delivery pipe and suction pipe on indicator diagram



pressure in cylinder

At beginning of suction stroke =  $H_{arm} - h_s - (H_{as})_{max}$

At end of suction stroke =  $H_{arm} - h_s + (H_{as})_{max}$

At beginning of delivery stroke =  $H_{arm} + h_d + (H_{ad})_{max}$

At end of delivery stroke =  $H_{arm} + h_d - (H_{ad})_{max}$

chances of cavitation are max. at beginning of suction stroke.

for no cavitation,  $H_{arm} - h_s - (H_{as})_{max} > H_{v.p}$

cavitation can also occur @ end of delivery stroke  $\rightarrow \omega_1 =$

for no cavitation

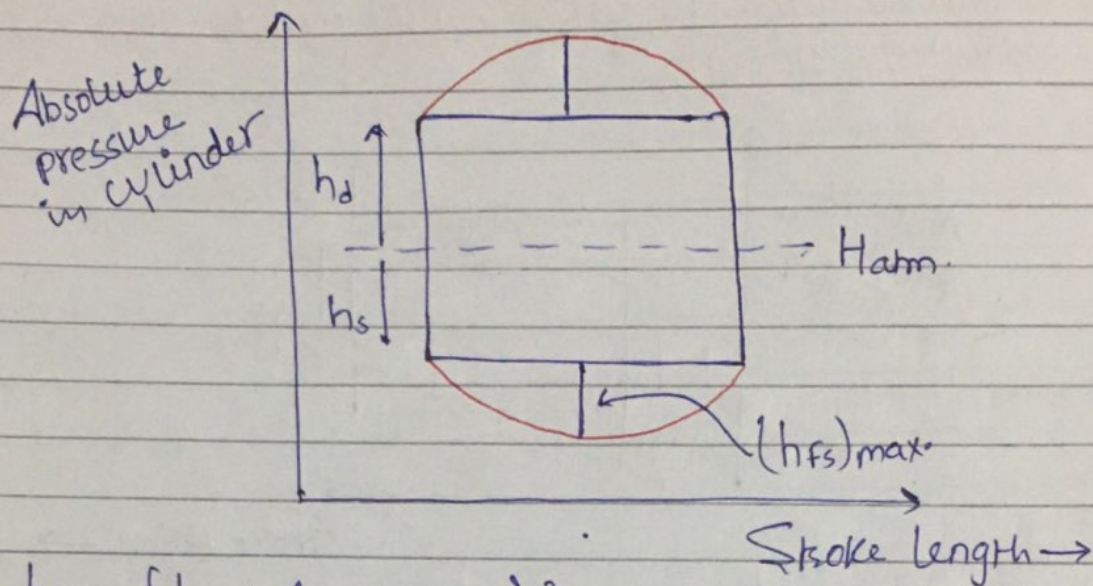
$H_{arm} + h_d - (H_{ad})_{max} > H_{v.p} \Rightarrow \omega_2$

Safe speed =  $\min(\omega_1, \omega_2)$

NOTE There is no effect of acceleration head on power input in a reciprocating pump.  
Due to acceleration head chances of cavitation increases.



## Effect of friction in suction & Delivery pipe on indicator Diagram



$$h_f = \frac{f l}{2 g d} \left( \frac{A}{a} r \omega \sin \omega t \right)^2$$

Power input to pump =  $K \times$  Area of indicator diagram.

$$= K \cdot L (h_s + h_d) + \frac{2}{3} (h_{fs})_{\max} L$$

$$+ \frac{2}{3} (h_{fd})_{\max} L$$

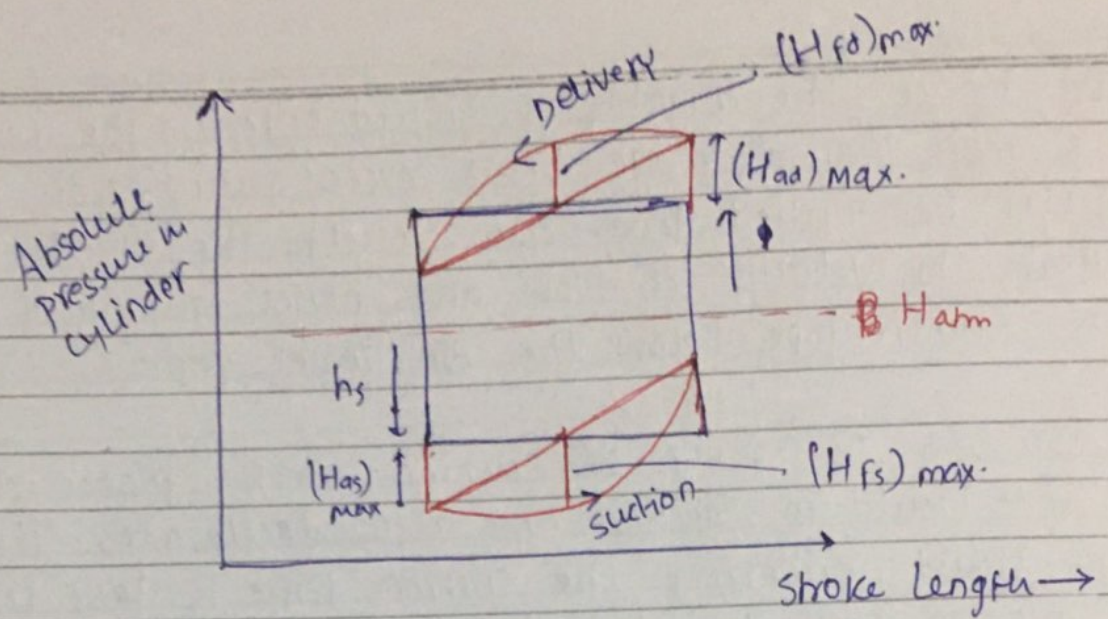
$$P_{i/p} = \rho g \left( \frac{A L N}{60} \right) \left( h_s + h_d + \frac{2}{3} (h_{fs})_{\max} + \frac{2}{3} (h_{fd})_{\max} \right)$$

\* Power ~~to~~ input to a reciprocating pump  $\uparrow$  due to friction in suction & delivery pipe

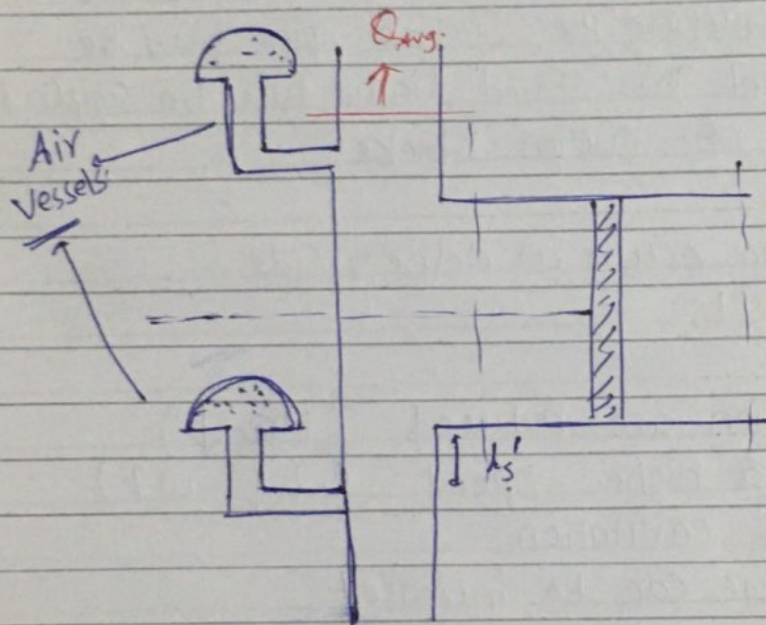
\* There is no effect of friction head on possibility of cavitation

## Effect of Acceleration and Friction in Suction and delivery pipe on indicator Diagram





### Air Vessels



$$H_{atm} - h_s - (H_{as})_{max} > H_{v.p}$$

$$(H_{as})_{max} = \frac{A}{a_s} \left( \frac{L_s}{g} \right) r \omega^2$$

Air vessel is a large closed chamber fitted in suction and delivery pipe close to the cylinder.

During the first half of suction stroke piston accelerates and hence liquid in suction pipe is accelerated



and hence the quantity of liquid entering the cylinder is more than average. This excess quantity of liquid will be supplied from air vessel to the cylinder so that the velocity of flow and hence quantity of water in suction pipe below the air vessel remains Uniform

During 2<sup>nd</sup> half of suction stroke piston discharges and liquid in suction pipe also decelerates. The quantity of liquid entering the suction pipe is less than average. Since liquid is flowing ~~for~~ with an avg. velocity in suction pipe below air vessel, excess quantity of liquid entering the suction pipe will be stored into air vessel. This stored liquid will be supplied during first half of suction stroke.

Similar phenomena occurs on delivery side.

### Functions of Air vessels

- ① chances of cavitation are reduced ( $H_{as} \downarrow$ )
- ② pump can be run @ higher speed ( $L_s \downarrow, W \uparrow$ ) without danger of cavitation.

Accordingly Discharge can be increased

$$Q = \frac{A L N}{60}$$

$$W \uparrow \Rightarrow N \uparrow \Rightarrow Q \uparrow$$

- ③ Suction head can be increased
- ④ Uniform Discharge can be obtained
- ⑤ power req. to drive the pump is reduced. because of reduction in power req. to overcome friction losses.

For single Acting pump % saving in work = 84.8 %  
Double Acting pump  $\rightarrow$  39.2 %



This power saved can be utilised to increase the delivery head.